# Boundary layers

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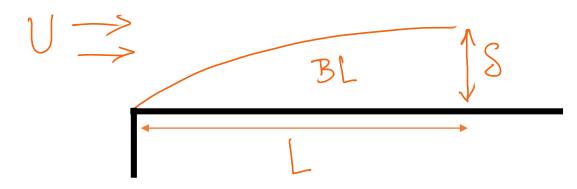
Part III Preparatory Workshop 2020

DAMTP, University of Cambridge

#### Motivation Bes: U, N = 0 but not uxn = 0 Outside BL: 4=70 (no-slip) Turbulent boundary layer Laminar boundary Stagnation point pressure = Total pressure p+

From: <a href="https://aerospaceengineeringblog.com/boundary-layer-separation-and-pressure-drag/">https://aerospaceengineeringblog.com/boundary-layer-separation-and-pressure-drag/</a> [accessed 26/09/2020] Original source unclear.

## Diffusion of vorticity



Kinematic viscosity  $\nu$ 

$$S \sim \sqrt{vt} \sim \sqrt{\frac{vL}{v}} \Rightarrow \frac{s}{L} \sim \sqrt{\frac{v}{LU}} = Re^{-\frac{1}{2}}$$

Mass conservation:

$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0$$

$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 6 \implies \frac{U}{L} \sim \frac{V}{S} \implies V \sim \Re^{-1/2} U$$

# Boundary layer equations (2D)

$$\chi: \rho\left(\frac{\partial u}{\partial t} + u\frac{\partial u}{\partial x} + v\frac{\partial u}{\partial y}\right) = -\frac{\partial p}{\partial x} + \mu\left(\frac{\partial^2 u}{\partial x^2} + \frac{\partial^2 u}{\partial y^2}\right)$$
same order
$$same \text{ order}$$

Outside BL: 
$$\rho\left(\frac{\partial U}{\partial t} + U\frac{\partial U}{\partial x}\right) = -\frac{\partial p}{\partial x}$$
 =>  $\phi \sim \rho V^2$ 

## Boundary layer equations (2D)

Recap: 
$$\frac{S}{L} \sim Re^{-1/2}$$
,  $\sqrt{\sim} \frac{S}{L} U$ ,  $\phi \sim \rho U^2$ 

$$y: \rho\left(\frac{\partial v}{\partial t} + u\frac{\partial v}{\partial x} + v\frac{\partial v}{\partial y}\right) = -\frac{\partial p}{\partial y} + \mu\left(\frac{\partial^2 v}{\partial x^2} + \frac{\partial^2 v}{\partial y^2}\right)$$

$$= > \frac{\partial p}{\partial y} = 0$$
Some order
$$\sim P^2$$

$$\frac{\partial \phi}{\partial y} = 0$$

$$1 \sim Re^{-1/2} \sqrt{1}$$

$$-\frac{\partial p_{\text{inBL}}}{\partial x} = -\frac{\partial p_{\text{outBL}}}{\partial x} = \rho \left( \frac{\partial U}{\partial t} + U \frac{\partial U}{\partial x} \right)$$

# Boundary layer equations (2D)

$$\rho\left(\frac{\partial u}{\partial t} + u\frac{\partial u}{\partial x} + v\frac{\partial u}{\partial y}\right) = \rho\left(\frac{\partial U}{\partial t} + U\frac{\partial U}{\partial x}\right) + \mu\frac{\partial u}{\partial y^2}$$

$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0$$

$$\frac{\partial v}{\partial x} + \frac{\partial v}{\partial y} = 0$$

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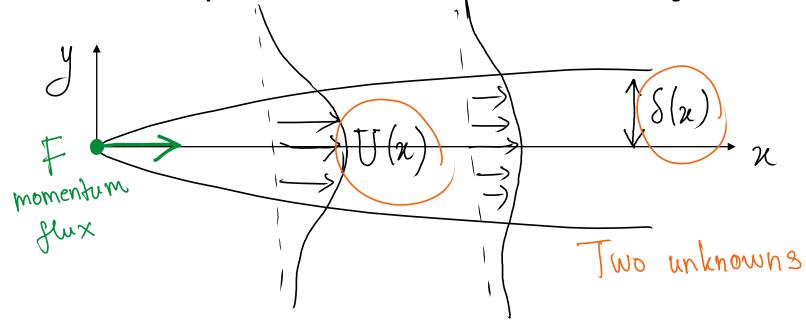
$$\frac{\partial v}{\partial x} + \frac{\partial v}{\partial y} = 0$$

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Boundary conditions:

$$u, v = 0$$
 on  $y = 0$   
 $u \to U$  as  $\frac{y}{\delta} \to \infty$ 

Example: 2D momentum jet



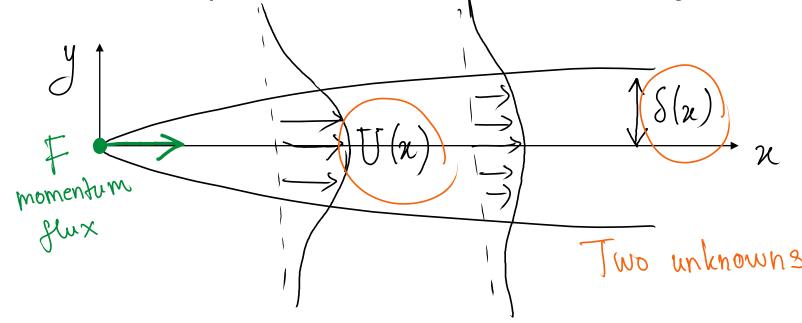
Steady 
$$\Rightarrow \frac{\partial u}{\partial t} = 0$$

No flow at infinity 
$$\Rightarrow \frac{\partial b}{\partial x} \Big|_{out} = 0$$

$$\rho \left( u \frac{\partial u}{\partial x} + v \frac{\partial u}{\partial y} \right) = \mu \frac{\partial u}{\partial y^2}$$
$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0$$

$$v = 0 \quad \text{on } y = 0$$
 $u \to 0 \quad \text{as } \frac{y}{\delta} \to \infty$ 

### Example: 2D momentum jet



$$\frac{\mathcal{T}^{2}(n)}{n} \sim \frac{\mu \mathcal{T}(n)}{S(n)^{2}}$$
 
$$S(n) = \left(\frac{\rho v^{2} n^{2}}{F}\right)^{\frac{1}{2}}$$

Momentum balance: 
$$\rho \mathcal{T}^{2}(n) \sim \mu \mathcal{T}(n)$$
  $\delta(n)^{2}$   $\delta(n)^{$ 

$$\rho \left( u \frac{\partial u}{\partial x} + v \frac{\partial u}{\partial y} \right) = \mu \frac{\partial u}{\partial y^2}$$

$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0$$

$$v = 0 \quad \text{on } y = 0$$

$$u = 0 \quad \text{on } y = 0$$

$$S(\chi) \sim \chi^{2/3}, \quad U(\chi) \sim \chi^{-\frac{1}{3}}$$

# Why constant momentum flux?

$$S \sim \chi^{2/3}$$

$$\Rightarrow \frac{\partial h}{\partial x} = -\frac{2}{3} \frac{h}{x}$$

Similarity solution

$$\delta(x) = \left(\frac{\rho \nu^2 x^2}{F}\right)^{1/3}, \quad U(x) = \left(\frac{F^2}{\rho^2 \nu x}\right)^{1/3}$$

(1) Streamfunction  $\psi$  s.t.  $u = \frac{\partial \psi}{\partial y}$ ,  $v = -\frac{\partial \psi}{\partial x}$ 

$$\psi(x,y) = \Psi(x,\eta) = \underbrace{U(x)\delta(x)}_{\text{s.t.}} f(\eta), \quad \eta = \frac{y}{\delta(x)}$$

$$\underbrace{\forall \psi(x,y) = \Psi(x,\eta) = \underbrace{U(x)\delta(x)}_{\text{s.t.}} f(\eta), \quad \eta = \frac{y}{\delta(x)}}_{\text{s.t.}} \sim \underbrace{U(u)}_{\text{as required}}$$

Sub into (x):  $\frac{\partial}{\partial y} = \frac{1}{8} \frac{\partial}{\partial y}$ ,  $\frac{\partial}{\partial x} = \frac{\partial}{\partial x} \left( -\frac{2}{3} \frac{y}{2x} \right) \frac{\partial}{\partial y}$ 

$$\Rightarrow u = \frac{\partial v}{\partial y} = U(u)f'(\eta), v = \frac{US}{x} \left(-\frac{1}{3}f + \frac{2}{3}\eta f'\right)$$

$$\rho \left( u \frac{\partial u}{\partial x} + v \frac{\partial u}{\partial y} \right) = \mu \frac{\partial u}{\partial y^2}$$

$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0$$

$$v = 0$$
 on  $y = 0$   
 $u \to 0$  as  $\frac{y}{\delta} \to \infty$ 

#### Similarity solution

$$\delta(x) = \left(\frac{\rho \nu^2 x^2}{F}\right)^{1/3}, \quad U(x) = \left(\frac{F^2}{\rho^2 \nu x}\right)^{1/3}$$

$$\psi(x,y) = \Psi(x,\eta) = U(x)\delta(x)f(\eta), \quad \eta = \frac{y}{\delta(x)}$$

$$\Rightarrow -\frac{1}{3}f'^2 - \frac{1}{3}ff'' = f'''$$

$$(1) = \Rightarrow f(0) = 0$$

$$(2) \Rightarrow f' \rightarrow 0 \text{ as } \eta \rightarrow \pm \infty$$

(2) => 
$$f'$$
>0 as  $\eta$ >  $\pm \infty$   
(3); const. mom flux =>  $\int_{-\infty}^{\infty} (f')^2 d\eta = 1$ 

$$\rho \left( u \frac{\partial u}{\partial x} + v \frac{\partial u}{\partial y} \right) = \mu \frac{\partial u}{\partial y^2}$$

$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0$$

$$v = 0$$
 on  $y = 0$  (1)  
 $u \to 0$  as  $\frac{y}{\delta} \to \infty$  (2)

#### Final remarks

 Remember to attempt the exercises for this topic before the live session on

#### 2pm Thursday, 8 October

• If you find any typos/mistakes in the PDF notes, please email me at mt599@cam.ac.uk

Thank you for watching!