# Recurrence of Travelling Waves in Transitional Pipe Flow

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The recent theoretical discovery of families of unstable travelling wave solutions in pipe flow (Faisst & Eckhardt 2003; Wedin & Kerswell 2004; Hof et al. 2004) at Reynolds numbers lower than the transitional range naturally raises the question of their relevance to the turbulent transition process. Here a series of numerical experiments are conducted in which we look for the spatial signature of these travelling waves in transitionary flows. Working within a periodic pipe of 5D (diameters) length, we find that travelling waves with low wall shear stresses (lower branch solutions) are on a surface in phase space which separates initial conditions which uneventfully relaminarise and those which lead to a turbulent evolution. This dividing surface (a separatrix if turbulence is a sustained state) is then minimally the union of the stable manifolds of all these travelling waves. Evidence for recurrent travelling wave visits is found in both 5 D and 10 D long periodic pipes but only for those travelling waves with low-to-intermediate wall shear stress and for less than about 10% of the time in turbulent flow at Re = 2400. Given this, it seems unlikely that the mean turbulent properties such as wall shear stress can be predicted as an expansion solely over the travelling waves in which their individual properties are appropriately weighted. Instead the onus is on isolating further dynamical structures such as periodic orbits and including them in any such expansion.

#### 1. Introduction

Wall-bounded shear flows are of tremendous practical importance yet their transition to turbulence is still poorly understood. Typically, the laminar flow solution is linearly stable (e.g. plane Couette flow, pipe flow) or if linearly unstable only well beyond the regime where transition occurs (e.g. channel flow). As a result, transition is an abrupt process triggered by a perturbation of sufficient amplitude. Generically, this could be expected to lead to an intermediate state of reduced symmetry but in fact the flow always immediately becomes temporally and spatially complicated. A recent new direction in rationalising this phenomenon revolves around identifying alternative solutions (beyond the laminar state) to the governing Navier-Stokes equations. In the past few years, such solutions in the form of steady states or travelling waves have been found in plane Couette flow (Nagata 1990; Clever & Busse 1992, 1997; Waleffe 1998), channel flow (Itano & Toh 2001; Waleffe 2001, 2003), an autonomous wall flow (Jimenez & Simens 2001) and most recently pipe flow (Faisst & Eckhardt 2003; Wedin & Kerswell 2004). Invariably, these solutions are saddle points in phase space. The idea is that through their stable and unstable manifolds and interconnections between them together with the appearance of periodic orbits, phase space becomes sufficiently complicated to support 'turbulent'

trajectories.

Gathering supporting evidence for this scenario is in its infancy, especially for spatiallyextended systems, but progress is being made. In reduced numerical models and spatiallyconfined systems, Eckhardt and coworkers have reported statistical evidence to suggest that these saddles collectively produce a chaotic repellor at least at low Reynolds numbers (Schmiegel & Eckhardt (1997); Schmiegel (1999); Eckhardt et al. (2002); Faisst & Eckhardt (2004); Eckhardt et al. (2007)). Other work has concentrated on establishing connections between specific flow behaviour and underlying nonlinear solutions present. In channel flow, for example, Itano and Toh (Itano & Toh 2001) interpret wall turbulent 'bursting' events with flow along the unstable manifold of a travelling wave solution. They also managed to isolate a periodic-looking solution on the basin boundary of the turbulence by continually adjusting a trajectory such that it neither relaminarised nor became turbulent (Toh & Itano 2003) (see Skufca et al. (2006) for an equivalent calculation in a model system). Jimenez et al. (Jimenez et al. 2005) studied both channel flow and plane Couette flow in an effort to relate near-wall turbulent events to the large number of known nonlinear solutions. They concluded that the turbulence stayed close to the upper branch travelling waves as far as comparing simple statistics of the flow field such as maximum (over space) wall-normal and streamwise components were concerned. Another approach has focussed upon identifying isolated periodic solutions directly from numerically-integrated turbulent trajectories using a Newton-Raphson technique. In both the case of plane Couette flow (Kawahara & Kida 2001) and highly-symmetric forced box turbulence (van Veen et al. 2006), the authors claim to find one periodic orbit which seems to share the same mean properties as the turbulent attractor.

In pipe flow, the only work so far aiming to establish the physical relevance of the recently-discovered travelling waves has been experimental (Hof et al. 2004, 2005). By analysing the flow structure across turbulent pipe flow (both of 'puff' and 'slug' type - see Wygnanski & Champagne (1973)), good correspondence was found, at least occasionally, with the outer symmetrically-arranged ring of fast 'streaks' (streamwise velocity anomalies) which is one of the dominant features of the travelling waves. The match is less clear, however, with regards to the complementary slow streaks centred around the pipe axis as well as with the smaller cross-stream velocities (e.g. figs 2E and 2F of Hof et al. (2004)). The purpose of this paper is to build on this work by carrying out a detailed quantitative study which can explore how closely the travelling waves are reproduced or 'visited' in phase space and the frequency of such visits using direct numerical simulations. The decision as to whether the flow has 'visited' a travelling wave is made by individually comparing the spatial structure of the instantaneous flow with that of each of the full set of travelling waves currently known to exist in the system. A concurrent numerical study by Schneider et al (2007) has chosen instead to focus on identifying coherent streak structures near the outer pipe wall by examining the azimuthal correlation of the downstream velocity there. Their results on how often the flow adopts a coherent streak structure near the wall provide a nice complementary upper estimate on how often we find travelling waves are visited. If it emerges that turbulent pipe flow can be understood as an effectively random switching between the neighbourhoods of these travelling waves then an appropriately weighted expansion across the 'active' travelling waves visited may provide a useful predictor of the turbulent flow properties. This presumes that some version of periodic orbit theory developed in low dimensional dynamical systems (e.g. (Cvitanovic 1988; Artuso et al. 1990a,b)) may carry over to this very high (formally infinite) dimensional setting.

The structure of the paper is as follows.  $\S 2$  begins by briefly describing the numerical method used to solve the Navier-Stokes equations before discussing reasonable measures chosen to quantify if, and how well, the flow approaches a travelling wave (TW) solution. There is a certain amount of arbitrariness in this choice because the TWs are fully nonlinear solutions not obviously orthogonal under any inner product. Hence some experimentation has been necessary before a final choice on the exact 'correlation' functions to evaluate made. Given also the fact that the TWs are parameterised continuously by their axial wavelength (albeit over a finite range), it has been convenient to impose a strict periodicity in the pipe to discretise the TWs which can exist in the system. As a result, a periodic pipe of length 5D has been used for the majority of the results. Even then, 37 TWs of two-, three- and four-fold rotational symmetry about the axis can be found at a Reynolds number of 2400. These are briefly described in  $\S 3$  together with their stability. In  $\S 4$ , we show numerical evidence that some of the TWs are visited but not all and for only part of the time. In  $\S 5$ , the statistical frequency of these visits is quantified by examining the correlation data from across a number of runs. Finally, a discussion follows in  $\S 6$ .

# 2. Formulation

#### 2.1. Numerics

The Navier–Stokes equation and solenoidal condition for the flow of an incompressible, Newtonian fluid along a circular, straight pipe under the action of an imposed pressure gradient are

$$\partial_t \mathbf{u} + \mathbf{u} \cdot \nabla \mathbf{u} + \frac{1}{\rho} \nabla p = \nu \nabla^2 \mathbf{u}, \qquad \nabla \cdot \mathbf{u} = 0,$$
 (2.1)

where  $\nu$  is the kinematic viscosity, p the pressure and  $\rho$  the constant density. Nondimensionalising the system using U the mean axial speed and the pipe radius D/2, where D is the diameter, gives rise to the Reynolds number  $Re := UD/\nu$ . A constant mass flow flow rate - or equivalently Re - is maintained along the pipe at all times. A numerical solution for the primitive variables (velocity and pressure) was developed in cylindrical coordinates  $(s, \theta, z)$  using finite differences in the radial direction (s) and Fourier modes for z and  $\theta$ . The time stepping was performed using the third-order Runge-Kutta scheme of Nikitin (2006). Hereafter a quoted numerical resolution of (N, M, K) corresponds to N+1 equally-spaced radial points (i.e. a grid step of 1/N where  $0 \le s \le 1$ ) and Fourier expansions in  $\theta$  and z of wavenumbers -M/2,...,M/2 and  $(-K/2,...,K/2)\pi/L$  respectively where L diameters is the nominal length over which periodicity is imposed. As discussed below, the main choice of pipe length was L=5: for this geometry a coarse grid was (25,32,30), an intermediate grid (50,48,40) and a fine grid (50,60,60): equivalent grids in longer pipes were used (e.g. a fine 10D-grid was (50,60,120)). Commonly in studies of this type the grid is stretched in the radial direction as the highest resolution is required near the wall. However, here there was little difference between the results obtained with uniform and non-uniform grids. This is not surprising given the nature of the flow at the Reynolds number considered in this study, as can be seen in Figure 3 below.

In addition to checks of specific components of the code using analytic test solutions, a series of calculations was performed using as initial conditions a TW solution plus a perturbation in the form of the leading (unstable) eigenfunction. Good agreement was

obtained for the growth of the disturbance and that predicted from the eigenvalue. Importantly, the code was also cross-validated with another time-stepping code based on velocity potentials (Willis & Kerswell 2007).

### 2.2. Travelling Waves

The TWs so far identified (Faisst & Eckhardt 2003; Wedin & Kerswell 2004) are arranged into symmetry classes of m-fold rotational symmetry about the axis and then continuously parametrised by their axial wavenumber across a finite range. TWs of 1-fold through to 6-fold symmetry have been found (Kerswell 2005) but only 2-, 3- and 4-fold TWs are currently known to exist below Re = 2485. Within each symmetry class, the TWs appear through saddle node bifurcations so that close to the saddle node point there is a well-defined upper and lower branch solution for a given wavenumber. For higher Re, the solution surfaces typically kink and fold back on themselves so that multiple pairs of branches can coexist at the same wavenumber (e.g. figure 10 of Wedin & Kerswell (2004)).

Imposing a pipe periodicity immediately reduces the continuum of TWs present down to a discrete number which can fit into the pipe. This is a crucial simplification which means the matching procedure adopted below can monitor all the TWs available to the flow. The main choice of a 5D long pipe was a compromise between the need to keep the number of TWs to a manageable size (helped by a shorter pipe) and the need to have a dynamical system which could support turbulent behaviour at a value of Re where the TWs are fully resolvable (helped by a longer pipe). This pipe length has also been studied before (Eggels  $et\ al.\ 1994$ ; Faisst & Eckhardt 2004) and sustained turbulence predicted for Re > 2250 (Faisst & Eckhardt 2004). This short length, of course, precludes capturing turbulent spatiotemporal features such as 'puffs' (Wygnanski & Champagne 1973) which typically extend over 20D but does allow an examination of 'temporal' turbulence which, when triggered, fills the whole pipe.

Figures 1 and 2 show the result of tracing out all the 2-,3- and 4-fold solution branches in the friction factor-axial wavelength plane at Re=2400 where the friction factor (Schlichting 1968) is defined as

$$\Lambda := -\frac{1}{\rho} \frac{dp}{dz} / \frac{U^2}{2D} \tag{2.2}$$

with dp/dz being the mean pressure gradient ( $\Lambda_{lam} := 64/Re$  is the laminar value). The curves are similar but less contorted at Re = 2000 (Kerswell 2005) and Re = 2200 (not shown). The vertical dotted lines drawn at axial wavenumbers  $\alpha = 0.625n$  where n = 1, 2, ...5 indicate the TWs which fit in the pipe which is actually taken to be  $\pi/0.625 = 5.0265$  diameters long. The m-fold symmetry class, the letter label and wavenumber are used to identify the TWs in what follows. For example, the TW with wavenumber 1.25 and lowest friction factor in figure 1 is the  $3b_1.25$  TW. For this pipe geometry and Re, there are 37 TWs (6 2-fold, 22 3-fold and 9 4-fold rotationally symmetric TWs) which can be numerically resolved and used to match against the flow.

#### 2.3. Matching

As fully nonlinear solutions, the TWs do not possess any simple orthogonality with respect to an inner product. Therefore establishing when a directly numerically simulated flow,  $\mathbf{u}_{DNS}$ , approaches a chosen TW velocity field,  $\mathbf{u}_{TW}$ , is not a straightforward case

of projection. Given this, a number of ad-hoc 'correlation' functions were developed and tested to measure how close the flow comes momentarily to a TW. The basic approach was to construct an integral based on the velocity fields  $\mathbf{u}_{DNS}$  and  $\mathbf{u}_{TW}$  over one wavelength  $2\pi/\alpha$  of the TW near the middle of the pipe. The correlation functions found most useful and thereby adopted were a normalised inner product based on the total velocity fields,

$$I_{tot}(t) = \max_{\theta_0, z_0} \left[ \frac{\langle \mathbf{v}_{DNS}, \mathbf{v}_{TW} \rangle}{\sqrt{\langle \mathbf{v}_{DNS}, \mathbf{v}_{DNS} \rangle} \sqrt{\langle \mathbf{v}_{TW}, \mathbf{v}_{TW} \rangle}} \right], \tag{2.3}$$

once the mean profile of the TW

$$\overline{w}_{TW}(s) := \frac{\alpha}{4\pi^2} \int_0^{2\pi/\alpha} \int_0^{2\pi} w_{TW}(s, \theta, z) \ d\theta dz \tag{2.4}$$

had been subtracted from both velocity fields

$$\mathbf{v}_{DNS} := \mathbf{u}_{DNS}(s, \theta_0 + \theta, z_0 + z) - \overline{w}_{TW}(s)\hat{z}$$

$$\mathbf{v}_{TW} := \mathbf{u}_{TW}(s, \theta, z) - \overline{w}_{TW}(s)\hat{z},$$

and an inner product using the cross-stream velocity components

$$I_{uv}(t) = \max_{\theta_0, z_0} \left[ \frac{\langle \mathbf{u}_{DNS}^{\perp}, \mathbf{u}_{TW}^{\perp} \rangle}{\sqrt{\langle \mathbf{u}_{DNS}^{\perp}, \mathbf{u}_{DNS}^{\perp} \rangle} \sqrt{\langle \mathbf{u}_{TW}^{\perp}, \mathbf{u}_{TW}^{\perp} \rangle}} \right].$$
(2.5)

Here  $\mathbf{u}^{\perp} = (u, v, 0)$  is the cross-stream velocity part of  $\mathbf{u}$ , and

$$\langle \mathbf{u}_1, \mathbf{u}_2 \rangle := \frac{\alpha}{2\pi^2} \int_0^{2\pi/\alpha} \int_0^{2\pi} \int_0^1 \mathbf{u}_1 \cdot \mathbf{u}_2 \, sds \, d\theta \, dz \tag{2.6}$$

is the (usual) inner product. The phase optimization over  $\theta_0$  and  $z_0$  is necessary because rotated and translated travelling waves are still solutions due to the symmetries of the flow.

One obvious alternative measure would be to use the velocity fields minus the laminar flow, as in commonly done in investigations of this type (including results presented below). However, it has major disadvantage it that it can produce large values of  $I_{tot}$ even when the flows do not match in any real sense. For example, Figure 13 below shows  $I_{tot}$  for a match to TW 2a\_1.25 for a run starting from TW 3b\_3.125. The 3-fold symmetry of the starting condition persists for some time (until  $t \approx 130$ ), and during this initial phase the correlation for the match to TW  $2a_{-}1.25$  with 2-fold symmetry is low (below 0.1), as might be expected. However, if the velocity minus the laminar value was used for the correlation, then  $I_{tot}$  would have a value of around 0.7 throughout this initial stage. Further, once the initial symmetry has broken, the match to 2a-1.25 produces values of  $I_{tot}$  of around 0.6, as shown in Figure 13. Using the laminar profile as a reference produces correlations of around 0.9. The relatively high value of the correlation coefficient during the initial phase even though the flows have different symmetries can be expained by the shape of the characteristic mean profile for the TW's. In common with turbulent flow in general, these show an increase in the streamwise velocity near the wall, and a decease in the centre of the pipe (see Wedin & Kerswell (2004) Figure 22) compared with the laminar profile. The high value of the correlation coefficient when using the laminar profile as the reference reflects this commonality in shape. Adopting the mean profile of the TW as reference avoids these misleadingly high correlations, but results in what may appear to be a relatively low value of  $I_{tot}$  even when there is a good match between the DNS flow and the TW.

However,  $I_{tot}$  is not in itself a sufficient measure for a match. Even when the mean profile of the TW has been subtracted, the streamwise (streak) velocities are typically an order of magnitude larger than the cross-stream velocities and hence their matching contribution tends to dominate  $I_{tot}$ , to the extent that there is little difference between  $I_{tot}$  and  $I_w$  (the equivalent of  $I_{uv}$  but using the streamwise velocity component only). Hence the introduction of the second measure  $I_{uv}$ .

By design,  $I_{tot}$  and  $I_{uv}$  can only take values in the interval [-1, 1] with a value 1 indicating a perfect match. The phase optimisation over  $\theta_0$  and  $z_0$  (carried out by systematically evaluating all the options over the  $[0, 2\pi) \times [-\pi/\alpha, \pi/\alpha)$  grid) in practice ensured that the correlations were never very negative, typically lying in the interval [-0.2, 0.2]. Experience indicated that there is evidence for a TW visit if  $I_{tot}$  and  $I_{uv}$  reach values of 0.5 and above (although more on this below).

A number of other measures were tried. An inner product using the streamwise vorticity, which provides a single measure of the cross stream flow, was found to shadow  $I_{uv}$ , although at a lower level. The symmetry assumed for the TWs (see (2.7) below) ensures that the cross stream velocity is zero along the axis of the pipe. This would not be expected to occur over an extended length of the pipe in fully turbulent flow. Hence an inner product over a subset of the domain excluding the central portion could be suitable. This was investigated, and it was found that restricting the domain to  $\frac{1}{2} \leq s \leq 1$  produced somewhat higher correlations, particularly for  $I_{uv}$ , but, again, a similar pattern of behaviour. Hence, these other measures could be used to produce essentially the same results by adjusting the level of correlation that would be regarded as giving a good match.

Correlation functions such as those introduced here measure shape only, and not amplitude. In theory, it would be possible to have a high correlation between velocity fields, but a significant difference in the values of the velocity. The value of the perturbation kinetic energy and the mean wall shear stress can be used to ensure that  $\mathbf{u}_{DNS}$  and  $\mathbf{u}_{TW}$  are similar, or to chose the best match when there is more than one candidate. Some of the TWs are highly correlated (values greater then 0.95 have been observed), so that, at a specific time, using these measures, there can be a good match of the flow to more than one TW. One such case will be considered below.

The matching was performed by maximising the value of  $I_{uv}$  over all possible values of  $\theta_0$  and  $z_0$ .  $I_{tot}$  was then calculated for the same orientation.  $I_{tot}$  is not suitable as the primary measure as it is so heavily dominated by the streamwise component that the cross-stream structure of the flow would, in effect, be discounted when choosing the "best" match.

#### 2.4. Travelling Wave Stability

An obvious way to start the *DNS* runs is to use a TW together with some small perturbation as an initial condition. If this perturbation is unstructured, for example, by relying on numerical discretization errors, the flow takes a long time to exit the neighbourhood of the TW and wastes cpu time. A better strategy is to find the unstable eigendirections of the TW and to use the most unstable eigenfunction with a small amplitude as the perturbation. Four 'lower' branch TWs - 2b\_1.25, 3a\_2.5,3h\_2.5 and 4b\_3.125 - and four 'upper' branch TWs - 2a\_1.25, 3b\_3.125, 3j\_2.5 and 4c\_3.125 - were selected as starting TWs. The distinction between upper or lower branch solutions can be ambiguous when

| Branch | TW            | Number of Unstable<br>Eigenvalues | Largest Growth Rate<br>(in units of U/D) | $\begin{array}{c} \text{Resolution} \\ (\mathcal{M}, \mathcal{N}, \mathcal{L}) \end{array}$ |
|--------|---------------|-----------------------------------|--|---|
| Lower  | 2b_1.25       | 1r                                | $1.1 \times 10^{-1}$                     | (9,25,7)  |
|        | $3a\_2.5$     | 2r                                | $2.1 \times 10^{-1}$                     | (8,30,6)  |
|        | $3h_{-2.5}$   | 3r+2c                             | $1.7 \times 10^{-1}$                     | (8,30,6)  |
|        | $4b\_3.125$   | 1r+2c                             | $2.6 \times 10^{-1}$                     | (6,40,5)  |
| Upper  | 2a_1.25       | 4c                                | $6.6 \times 10^{-2}$                     | (9,25,8)  |
|        | $3b_{-}3.125$ | 2c                                | $3 \times 10^{-3}$                       | (8,30,5)  |
|        | $3j_{-2.5}$   | 6c                                | $1.7 \times 10^{-1}$                     | (8,30,6)  |
|        | $4c_{-}3.125$ | 6c                                | $3.7 \times 10^{-1}$                     | (6,40,5)  |

Table 1. The stability properties of typical upper and lower travelling waves at Re = 2400: r and c indicate the number of real and complex eigenvalues respectively. The resolution  $(\mathcal{M}, \mathcal{N}, \mathcal{L})$  is the same for the travelling waves and the stability calculation and indicates the azimuthal, radial and axial resolution respectively: see Wedin & Kerswell (2004) for details. The unstable eigenvalues all correspond to disturbances possessing the same shift-&-reflect symmetry as the travelling wave.

the solution surface is as convoluted as in figure 1. Here we consider that TWs with high friction factors are upper branch solutions and those with low friction factor are lower branch solutions. The 8 choices made represent extreme and therefore unambiguous examples under this categorisation.

The stability properties of the 8 chosen TWs are listed in Table 1. The travelling waves all possess the shift-&-reflect symmetry S

$$S: (s, \theta, z) \to (s, -\theta, z + \pi/\alpha), \qquad S: (u, v, w, p) \to (u, -v, w, p). \tag{2.7}$$

so permitted linear disturbances can either be partitioned into those symmetric or antisymmetric with respect to S. When checked the TWs were invariably stable to antisymmetric disturbances so Table 1 concentrates exclusively on the situation in the S-symmetric subspace. The number of unstable directions is strikingly small given the large degrees of freedom involved (e.g. O(15,000)) and the size of the growth rates  $O(0.1\ U/D)$  indicate inertial instabilities. Figure 3 shows the structure of the most unstable eigenfunctions for the lower branch TWs 2b-1.25 and 3a-2.5. The 3a-2.5 TW gives a particularly clear example of how an unstable eigenfunction is concentrated in the regions of maximum shear in the TW streak velocity.

# 3. Results

A series of runs were performed by taking as initial conditions each of the selected 8 upper and lower TWs perturbed by a small amount of their most unstable eigenfunction. Since this perturbation can be added or subtracted, 16 runs were in fact done. This protocol highlighted a fundamental difference between the upper and lower branch TWs. For all the lower branch TWs tested, starting the run in one sense along the TW's most unstable manifold invariably led to an uneventful gradual relaminarisation, whereas starting in the other sense always produced a turbulent evolution: see Figure 4. Both signs

of perturbation, in contrast, produced a turbulent trajectory for the upper branch TWs. This implies that the 4 lower branch TWs (and by implication other lower branch TWs) and their stable manifolds are part of a boundary dividing regions of phase space which lead to the two different types of behaviour. At least part of their unstable manifolds are normal to this surface either directed towards the laminar or turbulent states. The fact that lower branch solutions may be embedded in such a dividing surface has been suggested before (e.g. Waleffe (2003)) despite only one known example in channel flow (Itano & Toh (2001)). The results here provide a more systematic verification of this idea. The surface - formally a separatrix if the turbulence is a sustained state - is undoubtedly more than just a union of lower branch TWs and their stable manifolds. In plane Couette flow, for example, Kawahara (2005) has recently found a periodic orbit embedded in the separatrix.

In the current situation, this dividing surface has deliberately not been called a 'separatrix' as the second initial observation when doing these runs at Re = 2400 with a 5D pipe is that the turbulence is only transitory. In other words, the laminar state is still the global attractor at Re = 2400 in a 5D periodic pipe although the flow can experience a long but ultimately finite turbulent episode. Not surprisingly, the length of these turbulent transients seems to depend sensitively on the exact numerical resolution used. This is particularly true with regards to the azimuthal and axial resolution - reducing this too far can produce what looks to be sustained turbulence since the excitation of small scales and the ensuing enhanced dissipation which tends to relaminarise the flow are suppressed (Orszag & Kells (1980)). We observed that turbulence remains transient when reducing the resolution from the working resolution (50, 60, 60) to (25, 30, 30) or (50, 24, 24) but looks to be sustained at (50,16,16) over a time greater than 3,000D/U: the resolution of Faisst & Eckhardt (2004) which predicts sustained turbulence at Re = 2250 is somewhere in between these last two choices. The issue of exactly when (or indeed if!) pipe flow turbulence becomes sustained is an area of much current interest at the moment (Peixinho & Mullin 2006; Hof et al. 2006; Willis & Kerswell 2007; Lagha & Manneville 2007).

The two runs started around the lower branch TW  $4b\_3.125$  illustrate the general behaviour well. Figure 5 plots the disturbance kinetic energy (per unit mass)

$$KE := \frac{1}{2\pi L} \int_{0}^{2L} \int_{0}^{2\pi} \int_{0}^{1} \frac{1}{2} (\mathbf{u} - \mathbf{u}_{lam})^{2} s ds d\theta dz$$
 (3.1)

of the flow versus the mean wall shear stress

$$\tau := \frac{1}{2\pi LD} \int_0^{2L} \int_0^{2\pi} \rho \nu \frac{\partial w}{\partial s} \bigg|_{s=1} d\theta \, dz \qquad \left[ = -\frac{1}{8} \rho U^2 \Lambda = \frac{\Lambda}{\Lambda_{lam}} \left( -8\rho U^2 / Re \right) \right] \quad (3.2)$$

for the flow evolution together with all 37 of the TWs present. Strictly when comparing with values for a TW, the kinetic energy and mean wall shear should be calculated for the section of the pipe containing the best match. However, for a pipe of this length, flow structures tend to persist over the full length of the pipe, particularly in the near wall region containing the streaks. The wall shear stress depends only on the streamwise velocity, which also provides the dominant component of the kinetic energy perturbation. As a result, there is usually little difference in these quantities for the full pipe and a section corresponding to one of the travelling waves. Larger differences were observed in longer pipes.

For one sign of the eigenfunction perturbation, the flow tamely relaminarises while

for the other it executes a long turbulent transient. During this latter evolution, there is evidence of close visits to at least 5 TWs. The first (labelled '1' in figure 5) is to  $4f_{-3}.125$  and occurs during the early stages as the flow trajectory moves away from  $4b_{-}3.125$ : see figure 6. This quality of the match is extremely high suggesting that there may be a heteroclinic connection between the two TWs (see also figure 4 which shows the trajectory passing straight over the open circle representing  $4f_{-}3.125$ ). The fact that even at t=0 there is already a considerable correlation with  $4f_{-}3.125$  indicates that  $4b\_3.125$  is structurally similar to  $4f\_3.125$ , an observation which is also true for some other groupings of TWs within the same rotational symmetry class. Figure 6 also shows the correlation signal for  $4c_3.125$  which has a very close visit after 120 D/U (labelled '4'). Taken together, the correlation functions for the other 4-fold TWs indicate that the flow retains its 4-fold symmetry until about 130 D/U (this was verified by examining the transient solution), whereupon it switches to a predominantly 3-fold symmetry. Figure 7 shows this switch-over well via the correlation function  $I_{tot}$  for 3a.3.125. This TW is the best candidate for a close visit (labelled '5' at  $t \sim 250~D/U$ ) over all the 3-fold symmetric TWs. Examining the instantaneous flow field at point 5 - Figure 8 shows clear evidence of 3 equally-spaced fast streaks around the outside of the pipe like 3a\_3.125 which are fairly streamwise independent. Surprisingly, there is little similarity between the inner slow streak structures. The axial vorticity is a convenient but particularly discriminating way of probing the cross-stream velocity fields as it involves taking derivatives. Given the low value of  $I_{uv}$ , the poor comparison is expected, however there is evidence that the DNS flow and the TW have roughly the same wavelength of variation.

During the dominantly 4-fold symmetric phase, there is also evidence for visits to  $2b\_0.625$  and  $2a\_1.25$  (labelled '2' and '3') shown in figure 9. A further way to characterise the extent of all these visits is to compare the kinetic energy and wall shear stress of the flow across the length of the pipe used for matching with the values associated with the TW. This data is collected together in figure 5. The relative closeness of these points to the corresponding values for the TW (e.g. '5' and  $3a\_3.125$ ) is further supporting evidence of a visit.

Runs started at upper branch TWs with either sign of unstable eigenfunction lead to turbulent-looking trajectories. Figure 10 shows one of these runs starting with  $3b\_3.125$ . Just as in the case of  $4b\_3.125$ , the initial rotational symmetry of the flow lingers for a substantial time. During this phase, there is clear evidence of close visits to  $3b\_2.5$ ,  $3c\_2.5$  and  $3j\_2.5$  (see figure 11). The first visit is particularly significant as the structural overlap between the initial TW ( $3b\_3.125$ ) and the visited TW ( $3b\_2.5$ ) is low: by point '1', both  $I_{tot}$  and  $I_{uv}$  for  $3b\_2.5$  have risen from  $\approx 0.4$  to near 0.8 in just over 50 D/U. Figure 12, which compares the initial condition, the DNS flow at point '1' and the TW  $3b\_2.5$ , confirms that the flow makes a significant adjustment to match with the new TW.

After a time of about  $125\,D/U$  - see Figure 13 - the flow switches to being more 2-fold symmetric. The flow then successively visits  $2a\_1.25$  and  $2b\_1.875$  before relaminarising at a time 350 D/U. Point 4 gives a very good comparison between the the DNS flow and the TW  $2a\_1.25$  because the correlation functions  $I_{tot}$  and  $I_{uv}$  are simultaneously large within the comparison wavelength: see Figure 14 at  $z\approx 1.5$ . The velocity plots - see Figure 15 - are very similar even up to the same vortex in the top left quadrant of the cross-sectional snapshots.

In Figure 10 point 5 is flagged as a match to TW 2b\_1.875 as this TW has the largest

correlation to the flow at this point, with  $I_{tot}=0.63$  and  $I_{uv}=0.61$ . However, this is not the only TW that satisfies the criterion for a good match. Specifically, TW  $2a\_1.25$  has  $I_{tot}=0.58$  and  $I_{uv}=0.49$ . Also,  $2a\_1.25$  is closer to point 5 than  $2b\_1.875$  in Figure 10. That both of these TWs show a high correlation with the flow is not surprising as the TWs also show a high correlation (matching  $2b\_1.875$  to  $2a\_1.25$  gives  $I_{tot}=0.53$  and  $I_{uv}=0.62$ ). Figure 16 shows a comparison between the flow at point 5 and the two travelling waves. In general, Figure 16 suggests that there is better agreement between TW  $2a\_1.25$  and the flow than  $2b\_1.875$ , despite the fact that it has a lower correlation as measured by  $I_{tot}$  and  $I_{uv}$ . That this can occur is not unexpected since  $I_{tot}$  and  $I_{uv}$  are measures of shape and not amplitude. Figure 16 and the position of the TWs on Figure 10 suggests there is a better agreement in amplitude between  $2a\_1.25$  and the flow than with  $2b\_1.875$ .

No significant correlations were found with any 4-fold TWs as the correlation functions  $I_{tot} \leq 0.3$  and  $I_{uv} \leq 0.45$  throughout the evolution. This was a persistent finding in all the other runs where a 2-fold or 3-fold symmetric TW was used to initiate the flow. Even when the flow was started by a 4-fold symmetric TW, correlations with other 4-fold symmetric TWs would only be significant in the initial phase of the flow evolution where the seeding symmetry is still present. In contrast, certain 2-fold and 3-fold TWs were consistently visited regardless of how the flow was initiated. The fact that the 4-fold symmetric TWs have a higher wall-shear-stress-to-kinetic-energy ratio than the flow seems to adopt is indicated by figures 5 and 10 and equivalent plots for other starting TWs (not shown). The flow never seems to visit the part of phase space where the 4-fold TWs are unless specifically inserted whereas some of the 2-fold and 3-fold symmetric TWs are in the active part of phase space populated by turbulent trajectories. This is probably a feature of the transitional Reynolds number used here as evidence for 4-fold TWs has been found experimentally in fully turbulent flow at Re = 3000 (Hof et al. (2004)). It may be that the part of phase space populated by the turbulent dynamics expands to encompass some 4-fold symmetric waves by the time Re increases to this value.

Information about which TWs are visited in each of the 12 turbulent runs (8 starting from upper branch TWs and 4 from lower branch TWs) is summarised in Tables 2 and 3. The criterion used to indicate a visit is  $I_{tot} > \lambda$  and  $I_{tot} + I_{uv} > 2\lambda$  with  $\lambda = 0.5$  and  $\lambda = 0.6$  for a closer visit. There is a strong correlation between the rotational symmetry of the starting TW and the waves subsequently visited even when the initial transient - defined as the period when  $I_{tot}$  for the initial TW decreases (typically of O(30 D/U)in duration) - is not considered. For instance, runs 1,2,5 and 10 - see Table 2 - show no evidence of visits to TWs with different rotational symmetry. However, across this suite of runs, all the TWs are visited at least once except for the 3 TWs  $3a_{-1}.25$ ,  $3b_{-1}.25$ and 3c<sub>1</sub>.25. Table 3 shows the more discerning results of only considering times when the flow leaves the rotational symmetry class of the initial TW. Examples of when this occurs have already been discussed in run 4 (at about 130 D/U: see Figure 7) and in run 7 (at about 125 D/U; See Figure 13). This makes it particularly clear that the 4-fold rotationally symmetric TWs are never visited except when the run is specifically started near one of these TWs (runs 4, 11 and 12) whereas 2-fold and 3-fold symmetric TWs are visited regardless of the starting symmetry (e.g. runs 3, 6, 7, 9).

# 3.1. Frequency of Visits

Given the evidence above that the TWs are visited, the next question is how frequently. To answer this, 'visit' statistics were compiled across a number of different runs which exhibited turbulent episodes. Ideally, these should be composed from one very long - e.g.  $O(10,000\ D/U)$  - turbulent run but, as discussed already, none could be generated at this Re where the TWs are available. As a result, two strategies at quantifying the TW visit frequency were undertaken. The first involved piecing together all the turbulent episodes produced during the suite of 5D runs described above. The second was to generate longer turbulent data sets in a double-length 10D pipe; for relatively short pipes, the length of pipe appears to have a significant effect of the time turbulence is sustained for, as can be seen from Table 4. These latter runs were almost exclusively initiated by randomly-selected velocity fields taken from a long turbulent coarse-grid run: see Table 4. The coarse grid is too underresolved to reproduce the TWs accurately and hence calculate correlation functions directly but was perfectly adequate to initiate turbulence in finer grid runs.

For the 5D runs started at a TW, it was necessary to decide when the flow had left the neighbourhood of the TW and had started to behave in a turbulent fashion. Two different criteria were adopted for this. In the first (data set A), the start of the turbulent phase was taken as the time at which the correlation with the starting TW had reached a minimum. This turned out to be more stringent a criterion (i.e. produced a later time) than requiring that the flow trajectory merely enter the part of the  $KE-\tau$  plane occupied by the turbulent state. In the second (data set B), the start of the turbulent phase was taken as when the flow broke out of the initial rotational symmetry class of the starting TW (e.g. for run 4 this occurs at about 130 D/U: see Figure 7). In both cases the end of the turbulent phase was well estimated by when the flow trajectory left this 'turbulent'  $KE-\tau$  area. The subsequent relaminarisation phase was easily identified with both the perturbation kinetic energy and wall stress decaying monotonically down to their laminar values (e.g. see figures 5 and 10). These strategies produced a correlation data sets lasting 4,154 D/U (A) and 2,363 D/U (B). For each, a joint probability density function was then computed for  $I_{tot}$  and  $I_{uv}$  corresponding to the TW with largest  $I_{tot}$  at a given time: see figures 17 and 18. There is a clear positive correlation between  $I_{tot}$  and  $I_{uv}$ and significant evidence for recurrent TW visits. Comparing the two pdfs, there are a lot of close visits during the interval after the initial transient but before the rotational symmetry class of the flow changes. This is presumably the result of the flow percolating out of the specific TW region in which it is initially inserted. As a result, the pdf from data set A is likely to overestimate the visit frequency.

Using the criterion that a visit occurs if  $I_{tot} > \lambda$  &  $I_{tot} + I_{uv} > 2\lambda$ , the percentage visit time is plotted against the quality of the visit  $\lambda$  in Figure 19. To assess how close a visit needs to be - i.e. what  $\lambda$  is required - to be able to predict the instantaneous wall shear stress, Figures 20 and 21 plot the instantaneous DNS wall stress (over the matching wavelength) against the wall shear stress associated with the visited TW for three different values of  $\lambda$ . The choice of seeking maximal  $I_{tot}$  to identify the closest TW was taken to facilitate the comparison as this should provide the best match between the DNS streak structure at the pipe wall with that of the TW. Even with this, the correlation results for set A indicate that only  $\lambda \geq 0.7$  level visits are really good enough to start making predictions of the wall shear stress from the visited TW. Since there are no  $\lambda = 0.7$  visits in set B,  $\lambda = 0.6$  is used to indicate closer visits and better

stress correlation is evident at this level compared to the  $\lambda=0.5$  results. Since the abscissa is discretized over the 37  $\tau_{TW}$  values, each vertical strip of data in these figures indicates that that TW has been visited. Interestingly, Figure 20 shows a visit bias to TWs with larger wall stress ( $\leq -1.5*8\rho U^2/Re$ ) whereas Figure 21 is oppositely skewed ( $\geq -1.5*8\rho U^2/Re$ ) to lower wall stress TWs. A possible explanation for this is set A is dominated in a 2-to-1 ratio with runs started by upper branch TWs compared to lower branch TWs. In the initial adjustment phase where the flow trajectory gradually leaves the vicinity of the initial TW, the flow visits other TWs and these are more likely to be upper rather than lower branch solutions in the neighbourhood of an initial upper branch solution. The reason lower branch TWs feature more in set B may be because the turbulent episodes considered were not long enough to desensitise the visit statistics from the final relaminarisation process where the flow preferentially passes by lower branch TWs. This could indicate that efforts to remove this phase from the data but may not have been wholly successful.

Further runs were carried out in a  $2\pi/0.625~D~(\approx 10~D)$  pipe in the search for more sustained turbulent data. Extending the spatial domain by a factor of 2 allows a whole new set of TW wavelengths to fit into the pipe, roughly doubling the number of allowed TWs. The correlation calculations were not extended to encompass this enlarged set for two reasons. Firstly the new TWs are interspersed within the 5D set which already sample the available solutions well. Therefore the new TWs should have structures very similar to the existing TW set. Secondly, the increase in the numerical overhead: currently calculating the correlation functions takes around 50% of the cpu time. However to compensate for this omission, the visit frequencies based on half the admissible TWs could justifiably be doubled. Initial conditions were randomly selected from an apparently sustained coarse turbulent run and used in intermediate- and fine-grid runs: see Table 4. The statistics across the accumulated 5,865 D/U long intermediate grid turbulent data were similar to those across the accumulated  $4,780\,D/U$  long fine grid turbulent data so the sets were merged to give a one large data set over 10,000 D/U in duration. The joint pdf of  $(I_{tot}, I_{uv})$  shown in Figure 23 resembles that from 5D data set B as does the wall stress comparison during visits shown in Figure 24. Again, the most visited TWs look to be those with lower wall stresses and essentially the same subset of TWs are visited closely -  $2a_1.25$ ,  $2b_1.875$  and  $3b_2.5$  in 5D data compared to  $2a_1.25$ ,  $2b_1.875$ , 3b\_2.5 and 3c\_3.125 in the 10D data. Figure 22 shows an example of a visit to 2a\_1.25 during a typical turbulent episode in this suite of runs. Significantly, no 4-fold symmetric TWs were visited in the 10D runs reinforcing the observation made in the 5D work that these TWs are not in the same part of phase space as that populated by a turbulent flow.

Figure 19 brings together the results of the frequency analysis by comparing the visit percentage as a function of visit quality ( $\lambda$ ) for the three data sets. Doubling the 10D frequency data (as only half the possible TWs are considered) brings the observed visit frequency more into line with the results from the 5D data set B at  $\lambda=0.5$ . Taken together, they suggest that TWs are visited for approximately 10% of the time in turbulent pipe flow. Using  $I_{tot} > \lambda$  as the criterion for a visit would increase the frequency of visits, by, in effect, ignoring the cross-stream component of the flow and considering the streamwise (streak) structure only.

Figure 25 shows the perturbation kinetic energy and the mean wall shear stress for a 10D pipe for one of the intermediate and fine grid runs started from coarse grid turbulent data. This shows that the turbulent flow in this pipe is concentrated in a small patch of

the  $KE-\tau$  space, with both grids occupying the same region of the space. The other intermediate and fine grid turbulent runs are also concentrated in this region. Coarse grid turbulent runs occupy a somewhat different region of the  $KE-\tau$  space with lower values of both variables. Both the intermediate and fine grid have the same radial resolution (N=50). A test run was performed with the same grid in  $\theta$  and z as the intermediate grid but an increased resolution in s (N=75). Again, the flow occupied the same region in the  $KE-\tau$  space.

Figure 26 shows power spectra of the surplus kinetic energy and the wall shear stress for turbulent flow in a 10D pipe. The spectra were generated by taking the data for the four longest fine grid runs that were started from coarse grid turbulent data, with a sample time of  $782.1\ D/U$ . A Hanning window was applied, with the windowed data scaled so that the variance matches that of the original data, and the spectra were generated by averaging over the four samples. Interestingly, the spectra show a peak, corresponding to a period of approximately  $98\ D/U$  which suggests a periodic orbit embedded in the turbulent dynamics  $\dagger$ . This is currently under investigation.

# 4. Discussion

At this point it is worth cataloguing the achievements of this investigation.

- (a) Using the travelling waves already known as starting points, all related travelling wave solutions (TWs) of varying wavelengths and azimuthal symmetries which exist at Re = 2400 have been traced out (Figures 1 and 2). They naturally partition into 3 distinct classes of particular rotational symmetry about the axis 2-fold, 3-fold and 4-fold symmetric TWs and 37 in total fit into a  $\pi/0.625$  ( $\approx 5$ ) D periodic pipe.
- (b) The linear stability of four 'lower branch' TWs (TWs with some of the lowest wall stresses of the 37) and four 'upper branch' solutions (TWs with the some of the highest wall stresses of the 37) has been carried out. All are inertially unstable and therefore saddles in phase space with growth rates typically of the size  $O(0.1\,U/D)$  and all have very low dimensional unstable manifolds.
- (c) All the lower branch solutions considered appear to sit on a surface which separates initial conditions which uneventfully relaminarise and those which lead to a turbulent-looking evolution. The fact that lower branch solutions may be part of such a dividing surface has been suggested before but here we provide systematic evidence of multiple embeddings. The surface formally a separatrix if the turbulence is a sustained state then minimally appears a union of lower branch TWs and their stable manifolds. In contrast, initial conditions near all the upper branch solutions tested invariably become turbulent.
- (d) Turbulence in a 5D periodic pipe at Re=2400 may be long-lived but ultimately appears only transient. Turbulence also seems transient in a 10D pipe at Re=2400 but the transients have, on average, a longer life. The mean lifetime of a turbulent episode is sensitive to the numerical resolution used, with turbulence appearing to last longer, if
- † The intermediate grid data produced spectra consistent with those for the fine grid, while the coarse grid also produced a peak, but at a higher frequency.

not sustained, in underresolved calculations.

- (e) Clear numerical evidence is found that travelling waves are visited during transitional flows which corroborates the experimental observations of Hof et al. (2004). As in the experiments, the dominant recognisable feature of these visits is the appearance of an outer ring of equally-spaced fast streaks. The more variable slow streak structure towards the pipe axis is less clearly reproduced.
- (f) A number of different correlation functions were experimented with to measure how 'close' a given perturbation velocity field is to that of a TW. Two,  $I_{tot}$  and  $I_{uv}$ , were chosen and a visit criterion  $I_{tot} > \lambda$  and  $I_{tot} + I_{uv} > 2\lambda$  developed based on a 'quality of visit' parameter  $\lambda$ . After examining the velocity matches at various different levels of  $\lambda$ , a value of 0.5 was taken as indicating a 'visit' with values of 0.6 and higher indicating a 'close' visit. The correlations do not measure the amplitude of the perturbation velocity, and ancillary measures (the mean wall shear stress and the disturbance kinetic energy per unit mass) were used to verify the match in particular cases.
- (g) Turbulent data in both 5D and 10D pipes indicate that some 2-fold and 3-fold symmetric TWs are recurrently visited while others are not. In particular, no evidence was found for visits to 4-fold symmetric TWs at Re=2400. The visited TWs correspond to low-to-intermediate wall stress solutions which are embedded in the same part of phase space as the turbulent dynamics. Other TWs are clearly in very different phase space locations and hence are never visited except if the flow is specifically inserted there initially (e.g. all the 4-fold TWs). The fact that a 4-fold symmetric wave has been observed experimentally (Hof  $et\ al.\ (2004)$ ) albeit at Re=3000 suggests that by then the turbulent part of phase space has expanded to encompass some of the 4-fold TWs.
- (h) Based on the correlation functions,  $I_{tot}$  and  $I_{uv}$ , used and the visiting criterion adopted ( $\lambda=0.5$ ), numerical evidence suggests that travelling waves are only visited for about 10% of the time in turbulent pipe flow at Re=2400. These figures are broadly in line with the study by Schneider et al. (2007) which quantifies the frequency of coherent streak states near the pipe wall. They find coherent streak structures for 24% of the time at Re=2200 and 20% of the time at Re=2500 consistent with the thinking that travelling wave visits are a strict subset of their coherent streak states. Above and beyond issues surrounding how a 'visiting' episode is determined, the exact visiting frequency found during this work must also be qualified by the fact that other currently-unknown travelling waves may well exist which would increase this statistic. In fact, very recently a whole new family of asymmetric travelling waves has been found at Re=2400 (Pringle & Kerswell (2007)) which would, of course, increase the visit frequency.

The last finding answers the original motivating question for this study. The fact that a turbulent flow is apparently visiting a TW only 10% of the time implies that it is of limited use to view turbulence purely as the random switching between the neighbourhoods of TWs. From the perspective of, say, predicting the average turbulent wall stress, an appropriately weighted sum of all the relevant TW wall stresses seems unlikely to work given that so much time is spent away from the TWs in phase space. One possible way out of this conclusion is that other phase space objects such as periodic orbits perhaps glimpsed in figure 26 - need to be included in any such expansion. Alternatively, if the time spent away from coherent flow features is truly significant then some sort of

weighted mixture of statistics describing the coherent and incoherent flow phases seems inescapable.

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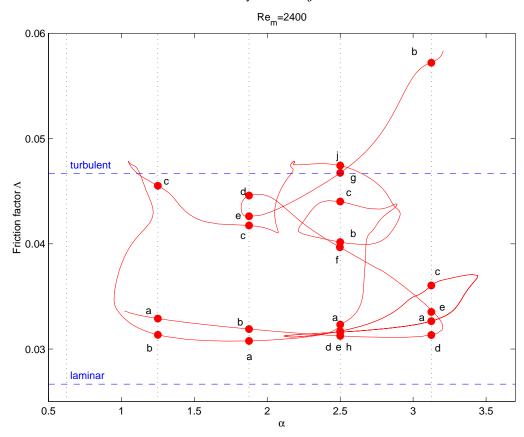


FIGURE 1. Solution branch for 3-fold rotationally-symmetric travelling waves plotted on a friction factor versus axial wavenumber plot at Re=2400. The lower blue dashed line represents the lower bound given by the Hagen-Poiseuille solution  $(\Lambda_{lam}=64/Re)$  and the upper blue dashed line corresponds to the Re=2400 value of the log-law parametrisation of experimental data  $\frac{1}{\sqrt{\Lambda}}=2.0\log(Re_m\sqrt{\Lambda})-0.8$  (see Schlichting 1968 equation (20.30)). The solution branch is only shown as far as it is assured resolved (hence the loose ends: the main mapping resolution was (8,30,6) in the truncation nomenclature of Wedin & Kerswell 2004). The dotted vertical lines indicate the wavenumbers ( $\alpha=0.625n$  in units of 2/D, n=1,2,3,4,5) which fit into a pipe of length  $\pi/0.625\,D$  long. The letters are used to label each allowable TW together with the wavenumber.

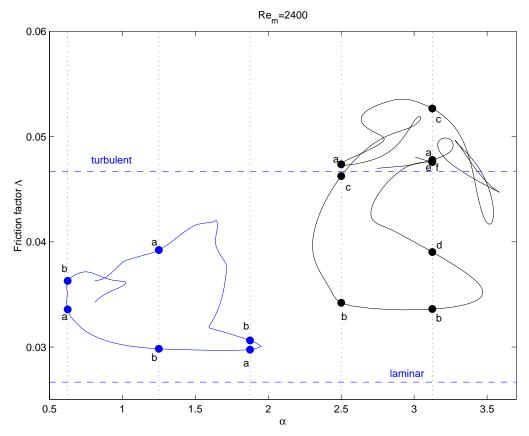


FIGURE 2. The equivalent of figure 1 but for the 2-fold (left,blue) and 4-fold (right,black) rotationally-symmetric travelling waves. Typical truncations used to resolve this solutions were (9,25,7) for the 2-fold case and (6,40,5) for the 4-fold case.

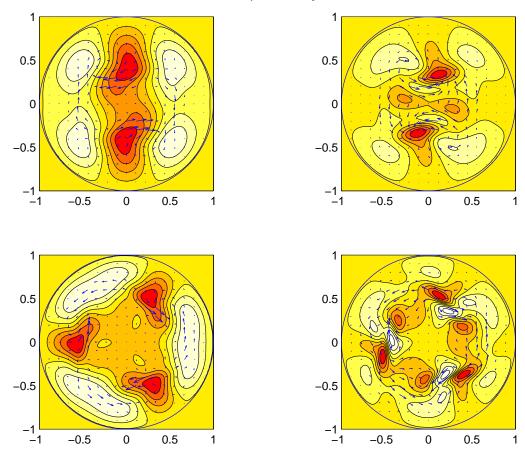


FIGURE 3. The travelling waves  $2b\_1.25$  (top left) and  $3a\_2.5$  (bottom left) and their most unstable eigenfunctions (top right for  $2b\_1.25$  and bottom right for  $3a\_2.5$ ). The arrows indicate the cross-stream velocities (larger arrows corresponding to larger speeds) for all 4 plots. The shading represents the axial velocity for the eigenfunctions whereas for the travelling waves, the axial velocity differential away from the laminar flow corresponding to the same mass flux is plotted (dark most negative - slow streaks - and light most positive - fast streaks). The same contour levels are used throughout with the eigenfunctions renormalised so that their largest axial velocity is set to the largest absolute contour level (the shading outside the pipe indicates 0: contours levels range from -0.417U to 0.266U in 8 steps).

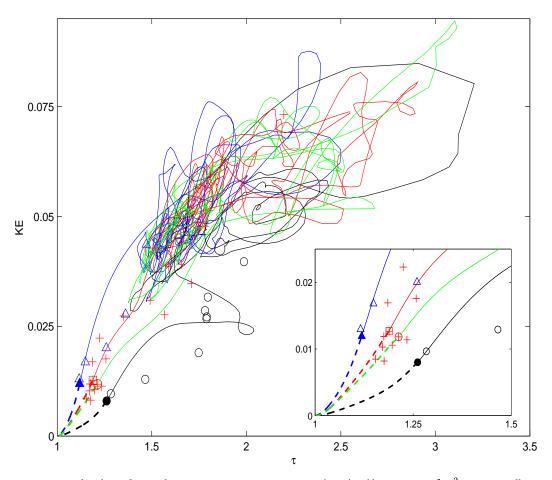


FIGURE 4. The disturbance kinetic energy per unit mass (see (3.1)), in units of  $U^2$  versus wall shear stress  $\tau$  (see (3.2)) in units of  $-8\rho U^2/Re$  for  $\mathbf{u}_{DNS}$  starting at the four lower branch TWs considered: 2b\_1.25 (marked as a solid triangle), 3a\_2.5 (marked as a square with a +), 3h\_2.5 (marked as a • with a +) and 4b\_3.125 (marked as a •). The solid line indicates the turbulent evolution for one sign of the eigenvalue perturbation for each TW and the thick dashed line traces out the uneventful relaminarisation for the other. The laminar state is represented by the point (1,0). All the TWs present are also plotted:  $\Delta$  for 2-fold TWs, + for 3-fold TWs and • for 4-fold TWs. The inset is a blow-up near the laminar point at (1,0) to highlight the relaminarisation.

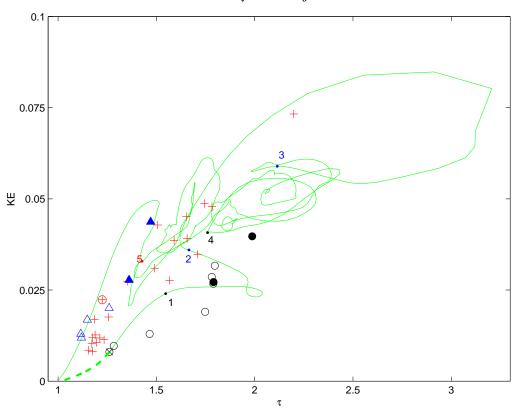


FIGURE 5. The disturbance kinetic energy per unit mass, in units of  $U^2$  versus wall shear stress  $\tau$  in units of  $-8\rho U^2/Re$  for  $\mathbf{u}_{DNS}$  starting at TW  $4b\_3.125$  (marked as a  $\circ$  with a  $\times$ ). The solid line indicates the turbulent evolution for one sign of the eigenvalue perturbation and the thick dashed line traces out the uneventful relaminarisation for the other (respectively  $4b\_3.125(+/-)$ ). The laminar state is represented by the point (1,0). All the TWs present are also plotted:  $\triangle$  for 2-fold TWs, + for 3-fold TWs and  $\circ$  for 4-fold TWs. Filled in symbols indicate TWs which appear to be visited by  $\mathbf{u}_{DNS}$  and the numbered dots correspond to the times of visits as determined by examining  $I_{tot}$  and  $I_{uv}$ . In chronological order:  $1 - 4f\_3.125$  (lower black filled circle);  $2 - 2b\_0.625$  (blue filled triangle furthest from 3);  $3 - 2a\_1.25$  (closer blue filled triangle);  $4 - 4c\_3.125$  (upper black filled circle) and  $5 - 3a\_3.125$  (circled red +).

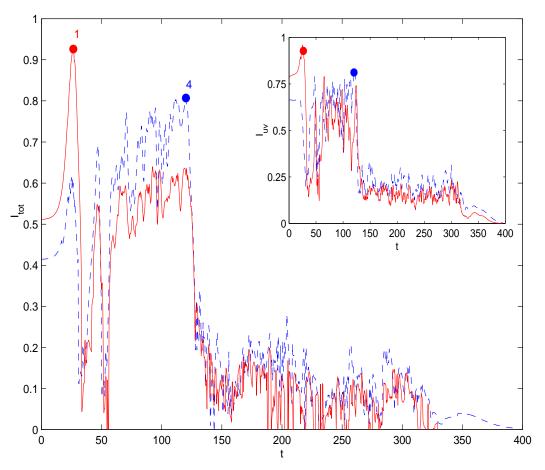


FIGURE 6.  $I_{tot}$  and  $I_{uv}$  (inset) as a function of time for TWs  $4f_{-}3.125$  (red solid line) and  $4c_{-}3.125$  (blue dashed line) starting at TW  $4b_{-}3.125$ . Dots label the times of likely closest visits to each TW (times '1' and '4' coincide with those in figure 5).

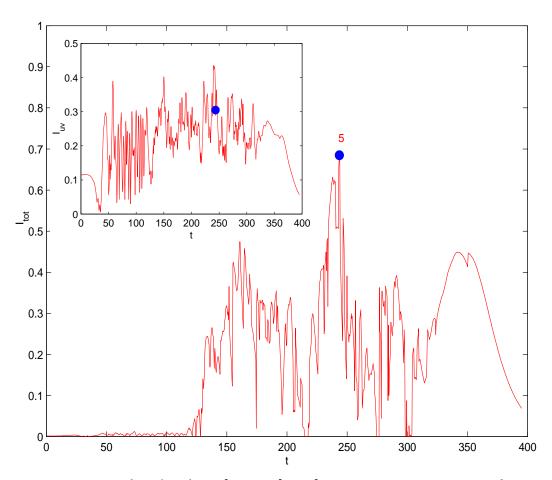


Figure 7.  $I_{tot}$  and  $I_{uv}$  (inset) as a function of time for TW  $3a\_3.125$  starting at TW  $4b\_3.125$ . The dot labels the time of closest visit (corresponding to '5' in figure 5).

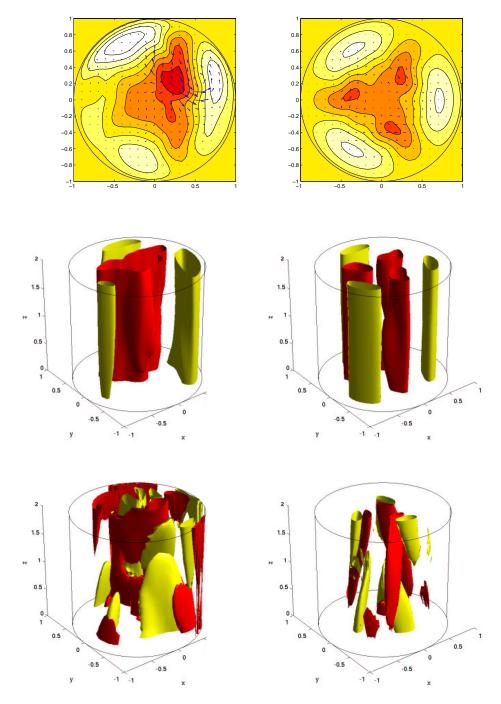


FIGURE 8. Comparison plots of the DNS flow (left column) and the TW  $3a\_3.125$  (right column) at point 5 in figure 7. The top row shows the velocity fields at the streamwise position of maximum  $I_{tot}(z)+I_{uv}(z)$  for a cross section of the pipe with the integration in the inner product performed over s and  $\theta$ . The shading represents the axial velocity perturbation from laminar flow with contours from -0.55 (dark) to 0.5 (light) for the DNS, and -0.4 to 0.35 for the TW, with a step of 0.15. The arrows indicate the cross stream velocity, scaled on magnitude (maximum 0.15 U). The middle row shows the streak structure over the wavelength of the TW, with contours of axial velocity at  $\pm 0.3\,U$  (light/dark). The bottom row shows the axial vorticity, with contours at  $\pm 0.6\,U/D$  (light/dark).

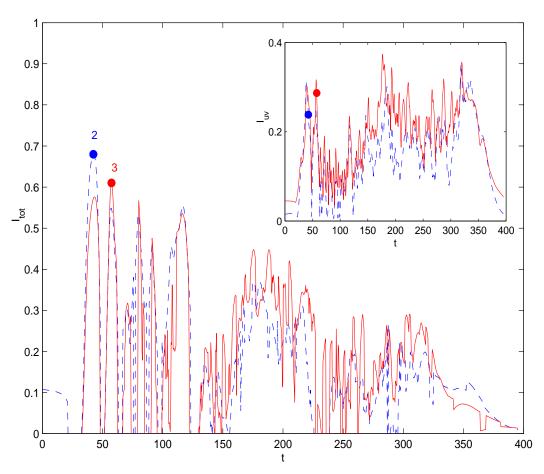


FIGURE 9.  $I_{tot}$  and  $I_{uv}$  (inset) as a function of time for TWs  $2b\_0.625$  (blue dashed line) and  $2a\_1.25$  (red solid line) starting at TW  $4b\_3.125$ . Dots label the times of likely closest visits to each TW (times '2' and '3' coincide with those in figure 5).

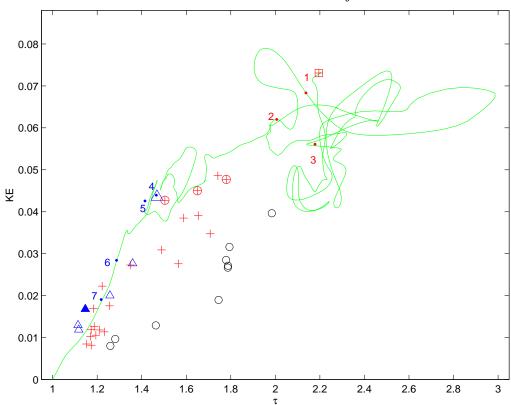


FIGURE 10. The disturbance kinetic energy per unit mass in units of  $U^2$  versus wall shear stress  $\tau$  in units of  $-8\rho U^2/Re$  for  $\mathbf{u}_{DNS}$  starting at the upper branch TW  $3b\_3.125(+)$  (marked as a  $\square$  with a + in it). The laminar state is represented by the point (1, 0). All the TWs present are also plotted:  $\Delta$  for 2-fold TWs, + for 3-fold TWs and  $\circ$  for 4-fold TWs. The numbered dots correspond to the times of visits as determined by examining  $I_{tot}$  and  $I_{uv}$ . In chronological order: 1 -  $3b\_2.5$  (rightmost red circled +); 2 -  $3j\_2.5$  (leftmost red circled +); 3 -  $3c\_2.5$  (middle red circled +); 4 and 6 -  $2a\_1.25$  (large blue triangle); 5 and 7 -  $2b\_1.875$  (blue solid triangle).

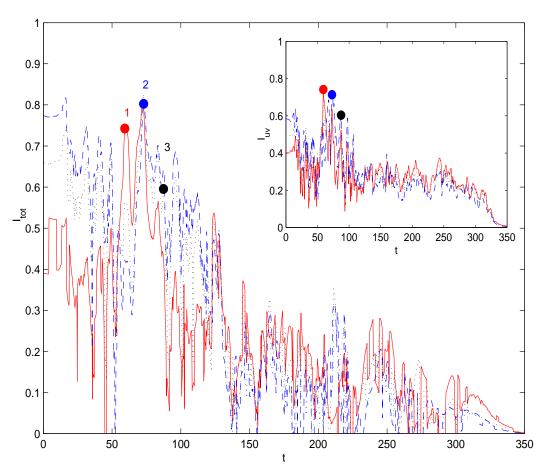


FIGURE 11.  $I_{tot}$  and  $I_{uv}$  (inset) as a function of time for TWs  $3b\_2.5$  (red solid line),  $3c\_2.5$  (black dotted line) and  $3j\_2.5$  (blue dashed line) starting at TW  $3b\_3.125$ . The dots label the time of closest visits (numbers correspond to points in figure 10).

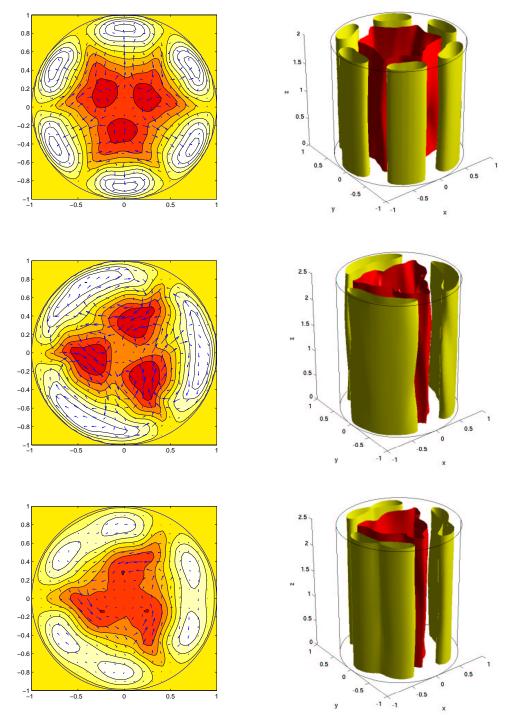


FIGURE 12. Comparison plots of the starting TW  $3b\_3.125$  (top), the DNS flow (middle) and the TW  $3b\_2.5$  (bottom) at point 1 in figure 11. The left column shows the velocity field at a slice where the DNS flow most matches  $3b\_2.5$ . The shading represents the axial velocity perturbation from laminar flow, with contours from -0.55 (dark) to 0.65 (light) (top), -0.55 to 0.5 (middle), and -0.55 to 0.35 (bottom), with a step of 0.15. The arrows indicate the cross stream velocity, scaled on magnitude (maximum 0.135 U). The right column shows the axial structure over a wavelength. Two contours of the axial velocity are shown at  $\pm 0.3 U$  (light/dark).

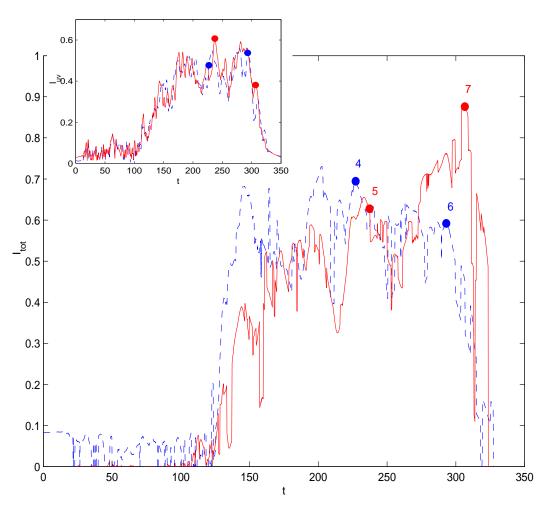


FIGURE 13.  $I_{tot}$  and  $I_{uv}$  (inset) as a function of time for TWs 2a\_1.25 (blue dashed line) and 2b\_1.875 (red solid line) starting at TW 3b\_3.125. The dots label the times of closest visits (numbers correspond to points in figure 10).

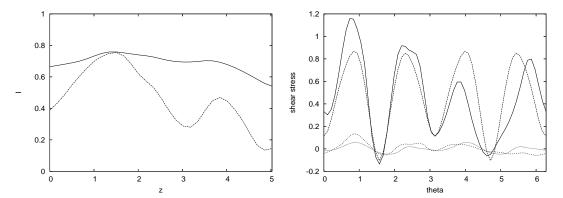


FIGURE 14. The left plot shows the correlations  $I_{tot}(z)$  (upper line) and  $I_{uv}(z)$  (lower line) over the wavelength of TW  $2a\_1.25$  with the highest correlation of  $I_{uv}$  at point 4 in figure 13.  $I_{tot}(z)$  and  $I_{uv}(z)$  give the correlations for a cross-section with integration in the inner product performed over s and  $\theta$ . The right plot shows the azimuthal distribution of the wall shear stress in units of  $-8\rho U^2/Re$  at the axial position of maximum  $I_{tot}(z)$  and  $I_{uv}(z)$  near z=1.5. From the top at the left of the plot the lines correspond to the axial stress from the DNS solution and that from the TW (minus the laminar value of one), and the azimuthal stress from the DNS solution and that from the TW.

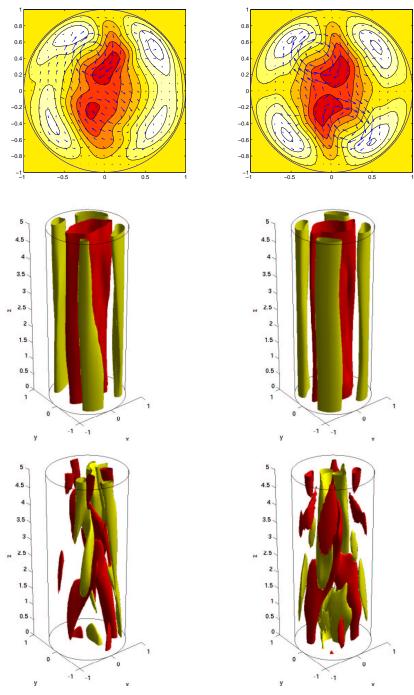


FIGURE 15. Comparison plots of the DNS flow (left column) and the TW 2a\_1.25 (right column) at point 4 in figure 13. The top row shows the velocity fields at the streamwise position of maximum  $I_{tot}$  shown in figure 14 ( $z \approx 1.5$ ). The shading represents the axial velocity perturbation from laminar flow with contours from -0.55 (dark) to 0.5 (light), with a step of 0.15. The arrows indicate the cross stream velocity, scaled on magnitude (maximum  $0.098\,U$ ). The middle row shows the streak structure over the wavelength of the TW, with contours of axial velocity at  $\pm 0.3\,U$  (light/dark). The bottom row shows the axial vorticity, with contours at  $\pm 0.8\,U/D$  (light/dark).

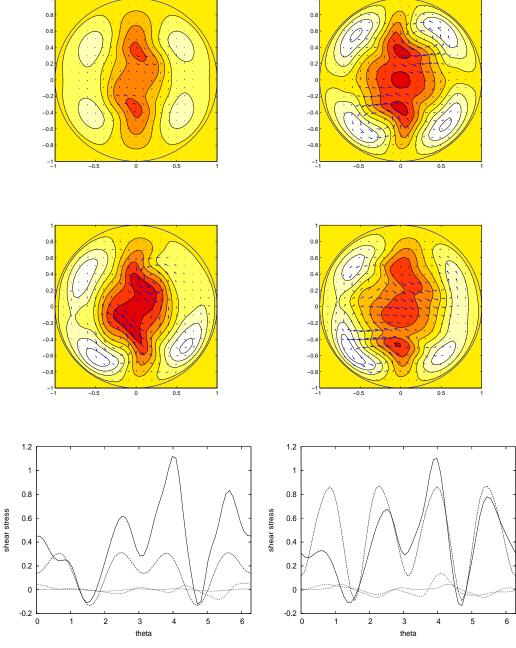


FIGURE 16. Comparison of the flow at point 5 in Figure 10 with TWs 2b-1.875 (left) and 2a-1.25 (right). The top row shows the velocity fields for the TWs at the streamwise position of maximum correlation over a cross section of the pipe, the middle row the DNS flow, and the bottom row the wall shear stress (in units of  $-8\rho U^2/Re$ ) at the same position. The shading represents the axial velocity perturbation from laminar flow with contours from -0.55 (dark) to 0.5 (light) (right column), -0.4 to 0.2 (top left), and -0.7 to 0.65 (middle left), with a step of 0.15. The arrows indicate the cross stream velocity, scaled on magnitude (maximum 0.162 U). For the wall shear stress, the top two lines show the axial stress (minus the laminar value of one). and the bottom two lines the azimuthal stress. For both the axial and azimuthal stress, the line with the largest peak is for the DNS solution, and the other (more regular) line from the TW.

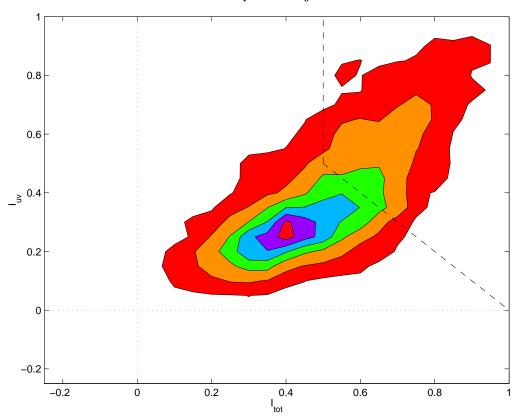


FIGURE 17. Joint pdf of  $I_{tot}$  and  $I_{uv}$  calculated over the turbulent episodes of all 5D runs started at the 8 upper and lower branch TWs (the values of  $I_{tot}$  and  $I_{uv}$  at a given time are selected by finding the TW with largest value of  $I_{tot}$ ) Contours are drawn at 0.01, 0.1, 0.3, 0.5, 0.7 and 0.9 of the maximum value. The dashed lines cordon off the visit region defined by  $I_{tot} > 0.5$  &  $I_{tot} + I_{uv} > 1$ .

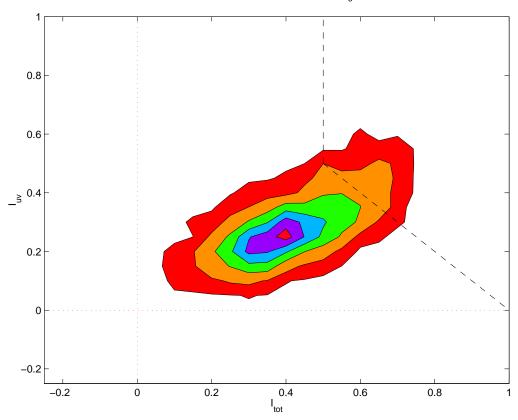


FIGURE 18. Joint pdf of  $I_{tot}$  and  $I_{uv}$  calculated over the turbulent episodes of all 5D runs started at the 8 upper and lower branch TWs using time episodes after the flow leaves the initial symmetry class (the values of  $I_{tot}$  and  $I_{uv}$  at a given time are selected by finding the TW with largest value of  $I_{tot}$ ) Contours are drawn at 0.01, 0.1, 0.3, 0.5, 0.7 and 0.9 of the maximum value. The dashed lines cordon off the visit region defined by  $I_{tot} > 0.5$  &  $I_{tot} + I_{uv} > 1$ .

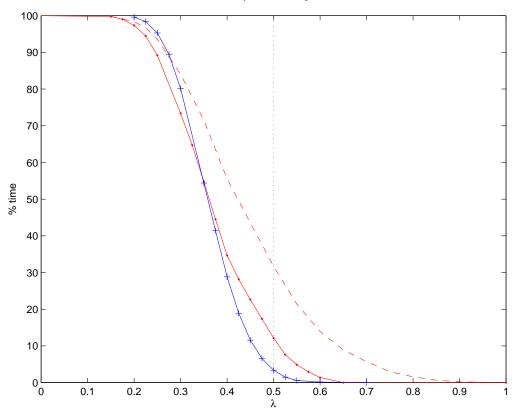


FIGURE 19. The percentage of time a turbulent flow 'visits' a TW based on the criterion  $I_{tot} > \lambda$  and  $I_{tot} + I_{uv} > 2\lambda$  as a function of  $\lambda$ . Statistics gathered from the 5D runs started at a TW with only the initial transient subtracted (dashed red line) and considering only times where the flow has left the initial symmetry class of the TW (solid red line with dots on), together with turbulent data compiled from various runs at 10D (blue solid line with crosses on).

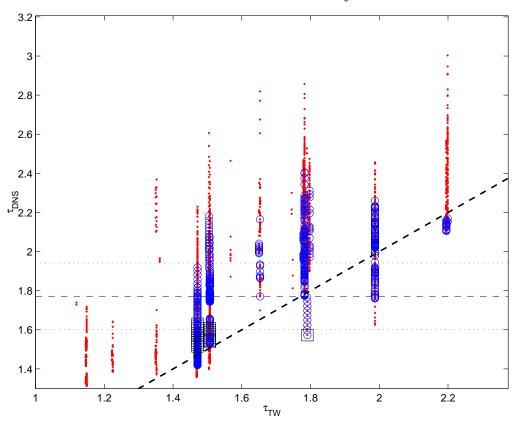


FIGURE 20. A plot of the wall shear stress  $\tau_{DNS}$  versus the wall shear stress  $\tau_{TW}$  of the TW being visited ( $\tau_{DNS}$  is measured across the matching wavelength of the TW) for the 5D data set A runs. The visit criterion is  $I_{tot} > \lambda$  &  $I_{tot} + I_{uv} > 2\lambda$ , where  $\lambda = 0.5$  (red dots) indicate approximate visits,  $\lambda = 0.7$  (blue circles) close visits and  $\lambda = 0.9$  (black squares) very close visits. The thick (black) diagonal dashed line indicates a perfect match  $\tau_{DNS} = \tau_{TW}$ . The horizontal (blue) dashed line is the mean wall shear stress across the whole pipe, the dotted lines indicate one standard deviation either side of this mean and the limits of the vertical axis have been set to the maximum (3.2) and minimum (1.3) wall shear stress values (in units of the laminar stress  $-8\rho U^2/Re$ ). Each vertical strip of points represents one TW - the close visits only occur for upper branch solutions where  $\tau_{TW} > 1.3$ .

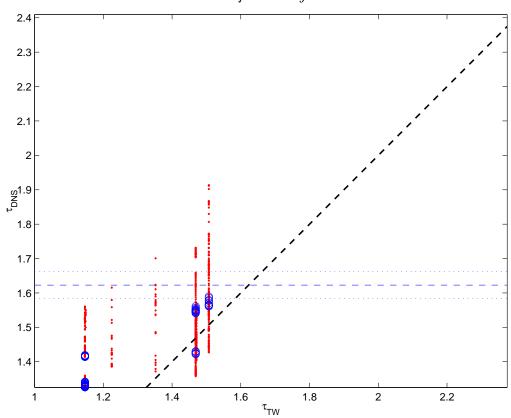


FIGURE 21. A plot of the wall shear stress  $\tau_{DNS}$  versus the wall shear stress  $\tau_{TW}$  of the TW being visited ( $\tau_{DNS}$  is measured across the matching wavelength of the TW) for the 5D data set B runs. The visit criterion is  $I_{tot} > \lambda$  &  $I_{tot} + I_{uv} > 2\lambda$ , where  $\lambda = 0.5$  (red dots) indicate approximate visits,  $\lambda = 0.6$  (blue circles) close visits. There are no very close ( $\lambda = 0.9$ ) visits. The thick (black) diagonal dashed line indicates a perfect match  $\tau_{DNS} = \tau_{TW}$ . The horizontal (blue) dashed line is the mean wall shear stress across the whole pipe, the dotted lines indicate one standard deviation either side of this mean and the limits of the vertical axis have been set to the maximum (2.4) and minimum (1.325) wall shear stress values (in units of the laminar wall stress  $-8\rho U^2/Re$ ). Each vertical strip of points represents one TW. For  $\lambda = 0.5$ , the TWs visited are  $2a_1.25$ ,  $2b_1.875$ ,  $3b_1.25$ ,  $3a_1.25$  and  $3c_1.25$ . The TWs closely visited ( $\lambda = 0.6$ ) are  $2a_1.25$ ,  $2b_1.875$  and  $3b_2.5$ .

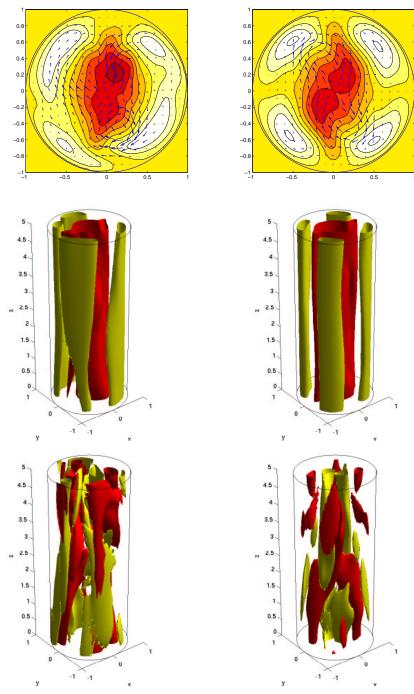


FIGURE 22. Comparison plots of the DNS flow (left column) and the TW  $2a\_1.25$  (right column) captured during a turbulent fine grid run in a 10D pipe. The top row shows the velocity fields at the streamwise position of maximum correlation over a cross section of the pipe. The shading represents the axial velocity perturbation from laminar flow with contours from -0.7 (dark) to 0.35 (light) for the DNS, and -0.55 to 0.5 for the TW, with a step of 0.15. The arrows indicate the cross stream velocity, scaled on magnitude (maximum 0.13 U). The middle row shows the streak structure over the wavelength of the TW, with contours of axial velocity at  $\pm 0.3 U$  (light/dark). The bottom row shows the axial vorticity, with contours at  $\pm 0.8 U/D$  (light/dark).

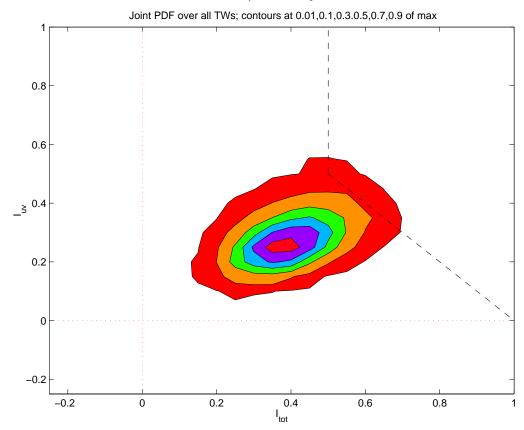


FIGURE 23. Joint pdf of  $I_{tot}$  and  $I_{uv}$  calculated over the turbulent episodes of all 10D runs (the values of  $I_{tot}$  and  $I_{uv}$  at a given time are selected by finding the TW with largest value of  $I_{tot}$ ). Contours are drawn at 0.01, 0.1, 0.3, 0.5, 0.7 and 0.9 of the maximum value. The dashed lines cordon off the visit region defined by  $I_{tot} > 0.5$  &  $I_{tot} + I_{uv} > 1$ .

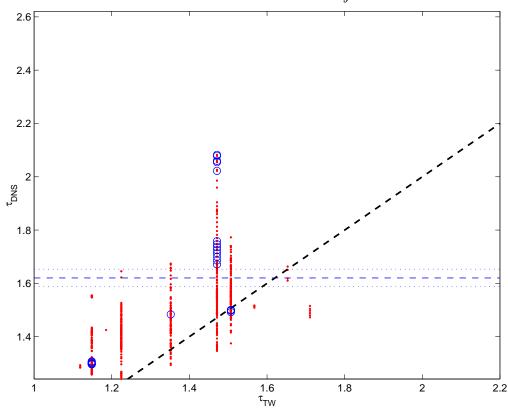


FIGURE 24. A plot of the wall shear stress  $\tau_{DNS}$  versus the wall shear stress  $\tau_{TW}$  of the TW being visited ( $\tau_{DNS}$  is measured across the matching wavelength of the TW) for the 10D runs. The visit criterion is  $I_{tot} > \lambda$  &  $I_{tot} + I_{uv} > 2\lambda$ , where  $\lambda = 0.5$  (red dots) indicates approximate visits and  $\lambda = 0.6$  (blue circles) closer visits (there are no  $\lambda = 0.7$  visits). The thick (black) diagonal dashed line indicates a perfect match  $\tau_{DNS} = \tau_{TW}$ . The horizontal (blue) dashed line is the mean wall shear stress across the whole pipe, the dotted lines indicate one standard deviation either side of this mean and the limits of the vertical axis have been set to the maximum (2.62) and minimum (1.24) wall shear stress values (in units of the laminar wall stress  $-8\rho U^2/Re$ ). Each vertical strip of points represents one TW. For  $\lambda = 0.5$ , the TWs visited are 2a.1.25, 2b.1.25, 3c.1.25, 2a.1.875, 3b.1.875, 3b.2.5, 3c.2.5, 3d.2.5, 3a.3.125 and 3c.3.125. The TWs closely visited ( $\lambda = 0.6$ ) are 2a.1.25, 2b.1.875, 3b.2.5 and 3c.3.125.

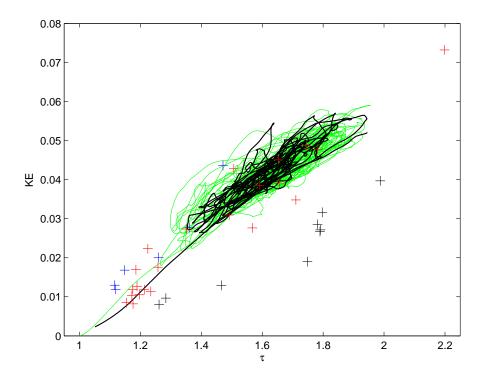


FIGURE 25. The disturbance kinetic energy per unit mass in units of  $U^2$  versus wall shear stress  $\tau$  in units of  $-8\rho U^2/Re$  for  $\mathbf{u}_{DNS}$  for two 10D runs started from a coarse grid turbulent run. The green line is for an intermediate (50,48,80) grid run and the black line a fine (50,60,120) grid run. The laminar state is represented by the point (1,0). All the TWs are also plotted: blue for 2-fold TWs, red for 3-fold TWs, and black for 4-fold TWs.

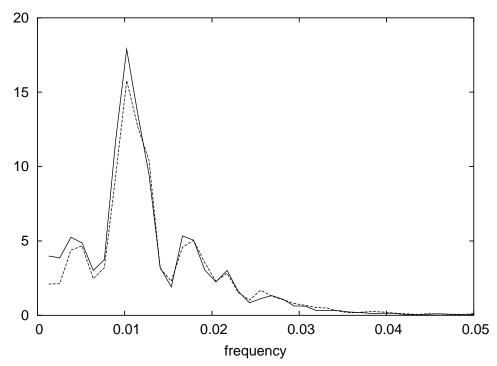


FIGURE 26. The power spectra of disturbance kinetic energy (multiplied by 5000) in units of  $U^3D$  (dashed) and wall shear stress in units of  $4UD/Re^2$  (solid) versus frequency. The maximum in the spectra corresponds to a period of approximately 98 in units of D/U.

| TW             | run | 1 | 2 | 3 | 4 | 5 | 6 | 7 | 8 | 9 | 10 | 11 | 12 |
|----------------|-----|---|---|---|---|---|---|---|---|---|----|----|----|
| 2a_0.625       |     | 0 |   |   |   |   |   |   |   |   |    |    | 0  |
| $2b\_0.625$    |     |   |   |   |   |   |   |   |   |   |    | 0  | 0  |
| $2a\_1.25$     |     | • |   | 0 |   | * | * | • |   | 0 |    |    |    |
| $2b\_1.25$     |     | * |   |   |   | • | • | 0 |   |   |    |    |    |
| $2a\_1.875$    |     | • |   | 0 |   | • | • | • |   |   |    |    |    |
| 2b_1.875       |     | • |   | 0 |   | • | • | • |   |   |    |    |    |
| 3a_1.25        |     |   |   |   |   |   |   |   |   |   |    |    |    |
| $3b\_1.25$     |     |   |   |   |   |   |   |   |   |   |    |    |    |
| $3c_{-}1.25$   |     |   |   |   |   |   |   |   |   |   |    |    |    |
| $3a\_1.875$    |     |   | • | 0 |   |   |   |   |   | 0 | 0  |    |    |
| $3b\_1.875$    |     |   |   |   |   |   |   |   |   |   | 0  |    |    |
| $3c\_1.875$    |     |   | • | • |   |   |   | 0 | 0 | • | •  |    |    |
| $3d\_1.875$    |     |   | 0 | 0 |   |   |   |   |   |   | 0  |    |    |
| $3e\_1.875$    |     |   | 0 | 0 |   |   |   | 0 |   | 0 | 0  |    |    |
| $3a\_2.5$      |     |   | * | • |   |   |   | 0 | • | 0 | 0  |    |    |
| $3b\_2.5$      |     |   | • | • |   |   | • | • | • | • | •  |    |    |
| $3c\_2.5$      |     |   | • | • |   |   | 0 | • | • | • | •  |    |    |
| $3d\_2.5$      |     |   | 0 | • | 0 |   | 0 | 0 | • | 0 | •  |    |    |
| $3e\_2.5$      |     |   |   |   |   |   |   | 0 |   |   | 0  |    |    |
| $3f_{-2.5}$    |     |   | • | • |   |   |   | • | • | 0 | •  |    |    |
| $3g\_2.5$      |     |   | • | • |   |   |   | • | • | 0 | •  |    |    |
| $3h_{-}2.5$    |     |   | • | * |   |   |   | 0 | 0 | 0 | 0  |    |    |
| $3j\_2.5$      |     |   | • | • |   |   |   | • | • | * | *  |    |    |
| $3a \_ 3.125$  |     |   | 0 | • | 0 |   | 0 | • | • | 0 | •  |    |    |
| $3b\_3.125$    |     |   | • | • |   |   |   | * | * | • | •  |    |    |
| $3c\_3.125$    |     |   | • | • | 0 |   | 0 | • | • | • | •  |    |    |
| $3d\_3.125$    |     |   |   | 0 |   |   |   |   |   |   |    |    |    |
| $3e\_3.125$    |     |   | 0 | 0 |   |   |   | 0 | 0 |   | 0  |    |    |
| $4a\_2.5$      |     |   |   |   | • |   |   |   |   |   |    | •  | •  |
| $4b\_2.5$      |     |   |   |   | 0 |   |   |   |   |   |    | 0  | 0  |
| $4c\_2.5$      |     |   |   |   | 0 |   |   |   |   |   |    |    | 0  |
| $4a\_3.125$    |     |   |   |   | • |   |   |   |   |   |    | •  | •  |
| $4b\_3.125$    |     |   |   |   | * |   |   |   |   |   |    | 0  |    |
| $4c$ _ $3.125$ |     |   |   |   | • |   |   |   |   |   |    | *  | *  |
| $4d\_3.125$    |     |   |   |   | • |   |   |   |   |   |    | •  | •  |
| $4e\_3.125$    |     |   |   |   | • |   |   |   |   |   |    | •  | •  |
| $4f\_3.125$    |     |   |   |   | • |   |   |   |   |   |    | •  | •  |

TABLE 2. Summary of visits for the 12 transitional runs (4 starting at lower branch TWs + 8 starting at upper branch TWs) which have turbulent episodes (data set A). TWs visited using the criterion  $I_{tot} \geq 0.5$  &  $I_{tot} + I_{uv} \geq 1$  at least once in a given run have the symbol  $\circ$  entered under that column against the (\* indicates the starting TW). A • indicates where the higher threshold of  $I_{tot} \geq 0.6$  &  $I_{tot} + I_{uv} \geq 1.2$  is satisfied. The initial transient as the flow moves away from the starting TW - typically of  $O(30\,D/U)$  duration - is not considered. The code for the run numbers is:  $1 - 2b\_1.25(-)$  (where (-) indicates the sign of the eigenfunction perturbation),  $2 - 3a\_2.5(+)$ ,  $3 - 3b\_2.5(+)$ ,  $4 - 4b\_3.125(+)$ , 5 &  $6 - 2a\_1.25(+/-)$ , 7 &  $8 - 3b\_3.125(+/-)$ , 9 &  $10 - 3j\_2.5(+/-)$ , 11 &  $12 - 4c\_3.125(+/-)$ .

| TW                 | run | 1    | 2              | 3    | 4    | 5              | 6         | 7 | 8    | 9    | 10   | 11 | 12 |
|--------------------|-----|------|----------------|------|------|----------------|-----------|---|------|------|------|----|----|
| 2a_0.625           |     | 0    | 0.83           |      |      | 0.86           |           |   | 0.86 |      |      |    |    |
| 2b_0.625           |     | 0.79 | 0.91           | _    |      | 0.91           |           |   | 0.87 | _    |      |    |    |
| 2a_1.25            |     | 0    | 0.01           | 0 04 |      | *              | *<br>0.95 | • | 0.89 | 0    |      |    |    |
| 2b_1.25            |     | *    | $0.81 \\ 0.91$ | 0.94 |      | 0.00           | 0.90      | • | 0.75 | 0.87 |      |    |    |
| 2a_1.875           |     |      | 0.91           | 0    |      | $0.88 \\ 0.92$ | 0         |   |      |      | 0.05 |    |    |
| 2 <i>b</i> _1.875  |     | 0    |                | O    |      | 0.92           | O         | • | 0.00 | 0.97 | 0.90 |    |    |
| 3a_1.25            |     |      |                |      |      |                |           |   |      |      |      |    |    |
| $3b_{-}1.25$       |     |      |                |      |      |                |           |   |      |      |      |    |    |
| $3c_{-}1.25$       |     |      |                |      |      |                |           |   |      |      |      |    |    |
| $3a\_1.875$        |     |      | 0.77           |      |      |                |           |   |      |      |      |    |    |
| $3b\_1.875$        |     |      | 0.11           |      |      |                |           |   |      |      |      |    |    |
| $3c_{-}1.875$      |     |      |                |      |      |                |           |   |      |      |      |    |    |
| $3d_{-}1.875$      |     |      |                |      |      |                |           |   |      |      |      |    |    |
| 3e_1.875           |     |      |                |      |      |                |           |   |      |      |      |    |    |
| $3a_{-}2.5$        |     |      | *              |      | 0.86 |                | 0.96      |   |      |      | 0.71 |    |    |
| $3b_{-2.5}$        |     | 0.89 | 0              | 0.91 | 0.98 |                | •         |   | 0.88 |      | 0.85 |    |    |
| $3c_{-2.5}$        |     | 0.00 | 0              | 0.0- | 0.87 | 0.85           | 0         |   | 0.00 |      | 0.00 |    |    |
| $3d\_2.5$          |     |      | 0              | 0.75 | 0    | 0.00           | 0         |   |      |      | 0.95 |    |    |
| $3e\_2.5$          |     |      |                |      |      |                |           |   |      |      |      |    |    |
| $3f_{-2.5}$        |     |      |                |      |      |                |           |   |      |      |      |    |    |
| $3g_{-2.5}$        |     |      |                |      |      |                | 0.73      |   |      |      |      |    |    |
| $3\ddot{h}_{-}2.5$ |     |      | 0.90           | *    | 0.83 |                | 0.93      |   |      |      |      |    |    |
| $3j\_2.5$          |     |      |                |      | 0.83 |                |           |   |      | *    | *    |    |    |
| $3a_{-3}.125$      |     |      | 0              | 0.88 | 0    | 0.88           | 0         |   |      |      | 0.97 |    |    |
| $3b\_3.125$        |     |      |                |      |      |                |           | * | *    |      |      |    |    |
| $3c\_3.125$        |     | 0.87 | 0              | 0.90 | 0    | 0.86           | 0         |   |      |      | 0.87 |    |    |
| $3d\_3.125$        |     |      |                |      |      |                |           |   |      |      |      |    |    |
| $3e\_3.125$        |     |      |                |      |      |                |           |   |      |      |      |    |    |
|                    |     |      |                |      |      |                |           |   |      |      |      |    |    |
| $4a\_2.5$          |     |      |                |      |      |                |           |   |      |      |      |    |    |
| $4b\_2.5$          |     |      |                |      |      |                |           |   |      |      |      |    |    |
| $4c\_2.5$          |     |      |                |      |      |                |           |   |      |      |      |    |    |
| $4a_{-}3.125$      |     |      |                |      |      |                |           |   |      |      |      |    |    |
| $4b\_3.125$        |     |      |                |      | *    |                |           |   |      |      |      |    |    |
| $4c\_3.125$        |     |      |                |      |      |                |           |   |      |      |      | *  | *  |
| $4d\_3.125$        |     |      |                |      |      |                |           |   |      |      |      |    |    |
| $4e\_3.125$        |     |      |                |      |      |                |           |   |      |      |      |    |    |
| $4f\_3.125$        |     |      |                |      |      |                |           |   |      |      |      |    |    |

Table 3. As for Table 2 except now only times when the flow has left the symmetry class of the starting TW are considered (data set B). Also numerical values are recorded of  $I_{tot} + I_{uv}$  in all near-miss instances where  $I_{tot} > 0.5$  but a  $\circ$  ( $\lambda = 0.5$ ) or  $\bullet$  ( $\lambda = 0.6$ ) is not warranted.

| Pipe Length | Run | Turbulent Duration in $D/U$ | Initiation       | $\begin{array}{c} \text{Resolution} \\ (N,M,K) \end{array}$ |
|-------------|-----|-----------------------------|------------------|---|
| 5 D         | 1   | 433 (673)                   | $2b\_1.25(-)$    | (50,60,60)  |
|             | 2   | $327\ (554)$                | $3a\_2.5(+)$     | (50,60,60)  |
|             | 3   | $340\ (517)$                | $3h_{-2.5(+)}$   | (50,60,60)  |
|             | 4   | $147\ (251)$                | $4b_{-}3.125(+)$ | (50,60,60)  |
|             | 5   | 195(385)                    | $2a_{-}1.25(+)$  | (50,60,60)  |
|             | 6   | 397 (609)                   | $2a_{-}1.25(-)$  | (50,60,60)  |
|             | 7   | $153\ (265)$                | $3b\_3.125(+)$   | (50,60,60)  |
|             | 8   | 263 (400)                   | $3b_{-}3.125(-)$ | (50,60,60)  |
|             | 9   | 90(257)                     | $3j_{-}2.5(+)$   | (50,60,60)  |
|             | 10  | 18 (133)                    | $3j_{-}2.5(-)$   | (50,60,60)  |
| 10D         | 1   | 879                         | ${f T}$          | (50,48,80)  |
|             | 2   | 1131                        | ${f T}$          | (50,48,80)  |
|             | 3   | 390                         | ${f T}$          | (50,48,80)  |
|             | 4   | 488                         | ${f T}$          | (50,48,80)  |
|             | 5   | 349                         | ${f T}$          | (50,48,80)  |
|             | 6   | 1446                        | ${f T}$          | (50,48,80)  |
|             | 7   | 1182                        | ${f T}$          | (50,48,80)  |
|             | 8   | 229                         | $2b\_1.25(-)$    | (50,60,120)   |
|             | 9   | 335                         | $\mathbf{T}$     | (50,60,120)   |
|             | 10  | 390                         | ${f T}$          | (50,60,120)   |
|             | 11  | 586                         | ${f T}$          | (50,60,120)   |
|             | 12  | 642                         | ${f T}$          | (50,60,120)   |
|             | 13  | 558                         | ${ m T}$         | $(50,\!60,\!120)$   |
|             | 14  | 265                         | ${f T}$          | $(50,\!60,\!120)$   |
|             | 15  | 640                         | ${ m T}$         | $(50,\!60,\!120)$   |
|             | 16  | 1135                        | ${ m T}$         | (50,60,120)   |

Table 4. This Table lists the various runs used to produce the 'visit' statistics. The 5D data comes from initiating the code with a disturbed TW and two 'turbulent' duration times are listed. The first is measured from when the flow leaves the initial symmetry class of the TW (used for data set B) whereas the second, larger figure (in parentheses) is based on when the correlation function for the starting TW stops decreasing which is a much weaker condition that occurs earlier (used for data set A). The 10D runs are almost exclusively started using randomly selected velocity fields from a long turbulent run generated using a coarse (25,32,60) grid (a strategy labelled as 'T' in the Table). Note that some of the 10D runs were still turbulent when this data was gathered, whereas all the 5D runs had relaminarised.